
**PASI 2010 in Dynamics and Control of Manned and Unmanned
Marine Vehicles**

Dynamics and Hydrodynamics of High Speed Craft

1400-1545, June 29, 2010
Barranquilla, Colombia

Armin Troesch
Naval Architecture and Marine Engineering
University of Michigan

OUTLINE

- Planing Hull Hydrodynamics and Dynamics
Selected experiments and theories
with practical applications
- Nonlinear Dynamics Analysis Applied to Planing Hulls
- Stochastic Vibro-Impact Model for Extreme Planing
Craft Acceleration Estimation



With a little help from friends:

Recently Active UM Graduates:

Dr. Lixin Xu

Dr. Carolyn Frank Judge

Wayne Arguin (USCG)

Timothy Conners (USCG)

Dr. Richard Royce

Tony Daniels

Dr. Kevin Maki

Dr. Brant Savander

Oscar Tuscan (Dr. to-be)

Fredy Zarate

And many more.....



Planing and Impact Research at UM

- Kang, K.G. (1988) Non-linear Impact Hydrodynamics. Ph.D. Thesis.
- Talmor, A. (1991) Non-linear Slender-Body Approach for Predicting Planing Performance, Ph.D. Thesis.
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- Wang, M. (1995) A Study on Non-linear Free Surface Flows. Ph.D. Thesis.
- Savander, B.R. (1997) Planing Hull Steady Hydrodynamics. Ph.D. Thesis.
- Xu, L. (1998) A Theory for Asymmetric Vessel Impact and Steady Planing. Ph.D. Thesis.
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- Royce, R.A. (2001) Impact Theory Extended to Planing Craft with Experimental Comparisons. Ph.D. Thesis.
- Daniels, A.S. (2002) Design and Construction of the Stepped Planing Hull Dynamometer. Professional Degree Thesis.
- Maki, K. (2005) Transom Stern Hydrodynamics, Ph.D. Thesis.
- Tascon, O. (2010) Numerical Computation of the Hydrodynamic Forces Acting on a Maneuvering Planing Hull via Slender Body Theory - SBT and 2-D CFD Impact Theory. Ph.D. Thesis
- Rose, C. J. (2010) Marine Systems Application of Extreme Value Prediction of a Vibro-Impact System Subject to Stochastic Excitation, MSE Thesis
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Selected University of Michigan References

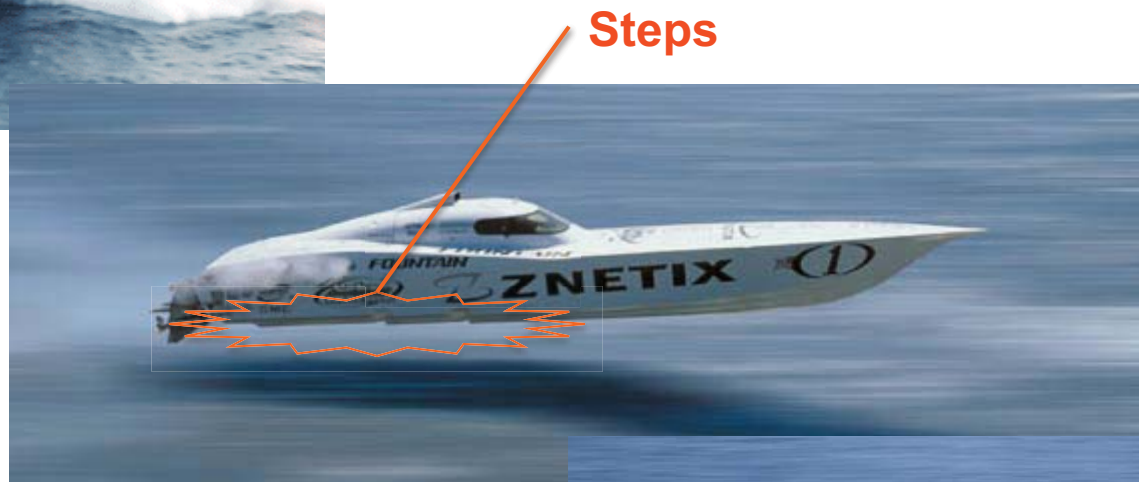
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Planing Craft Resistance and Dynamics



Planing Craft Resistance and Dynamics



Design & Analysis Process Overview

□ Steady (?) Hydrodynamic Analysis

- Defines global geometry
- Defines weight distribution
- Defines operating parameters

□ Seakeeping/ Maneuvering Analysis

- Defines structural loading
- Defines operability boundaries

□ Structures

- Quasi-Static & Transient
- Primary, Secondary, Tertiary
- Fatigue Life (Endurance Limit)
- Fracture Mechanics (Module Dis-assembly)

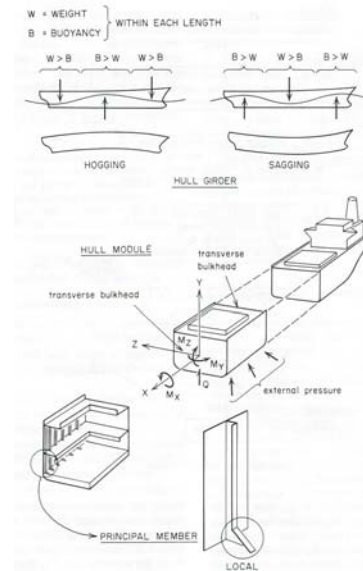
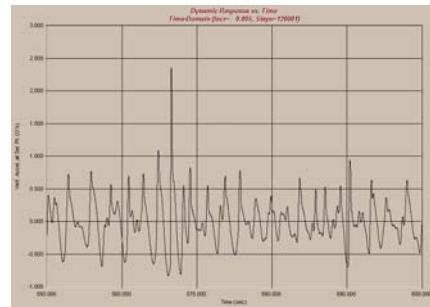
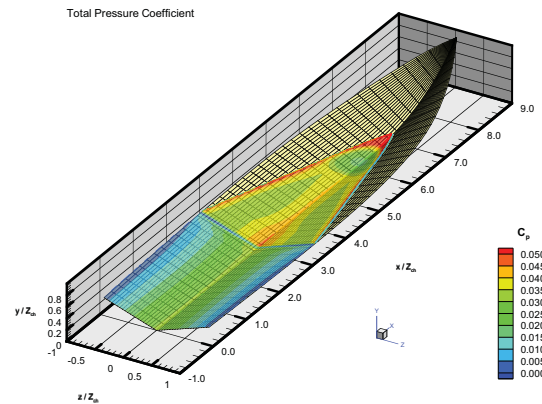
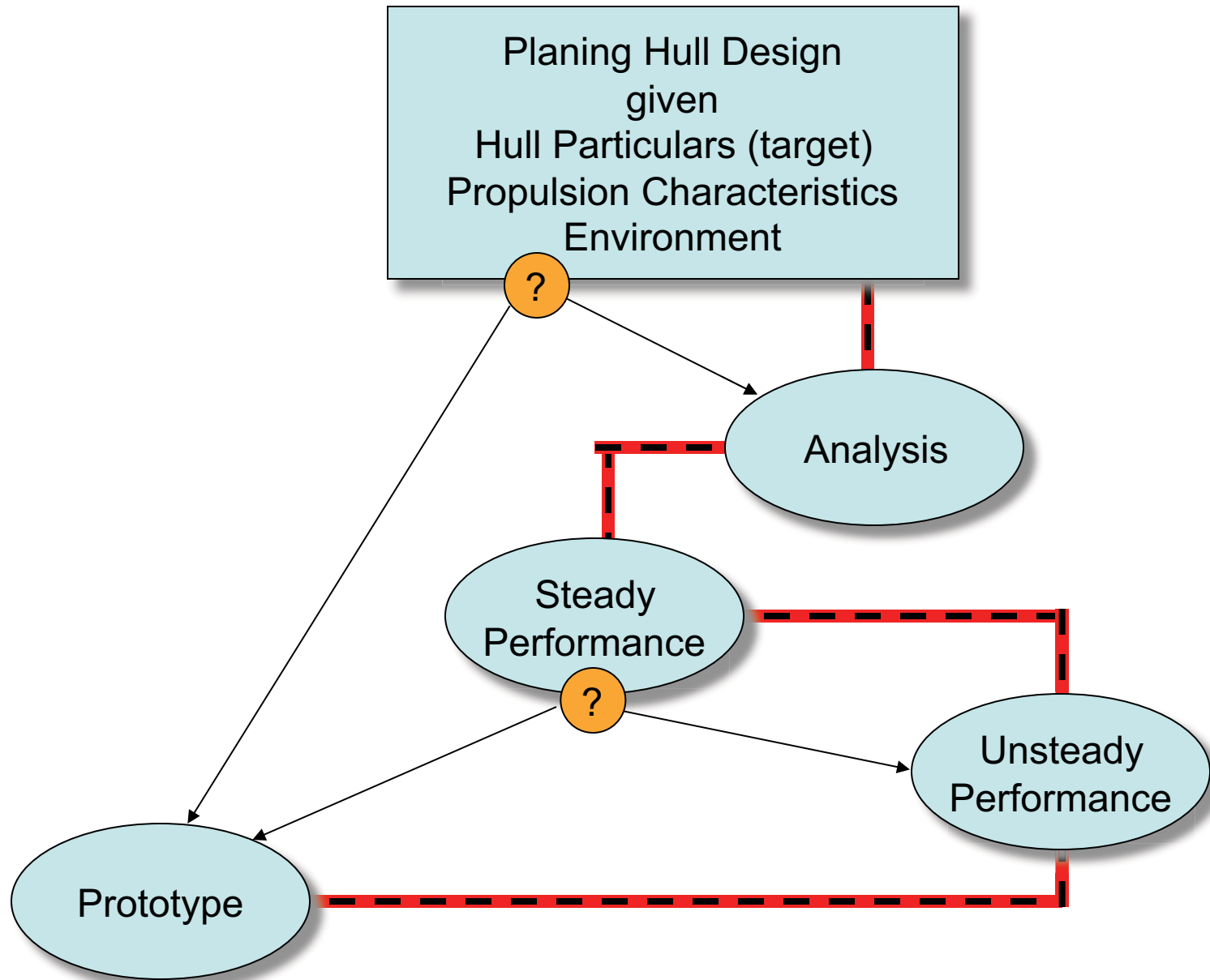


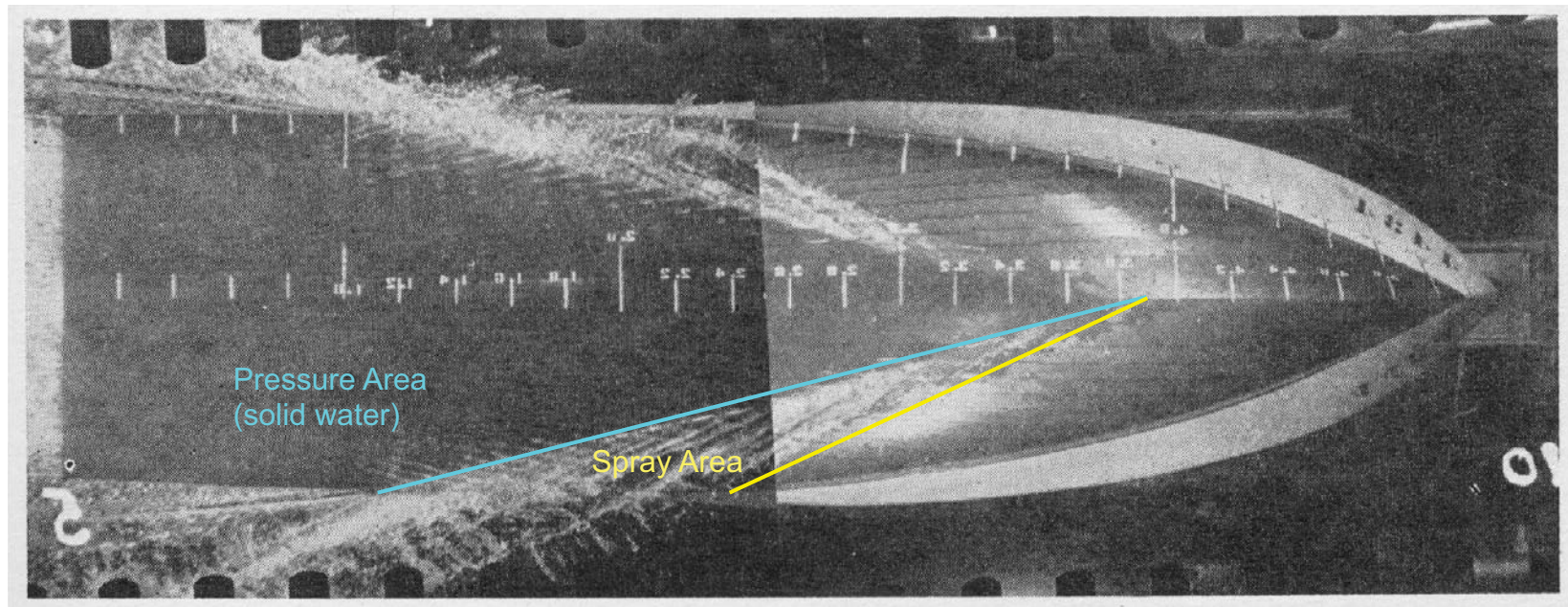
Figure 1.2 Levels of structural analysis.





Underwater Photograph of a Planing Hull

David Taylor Model Basin: Model 4434



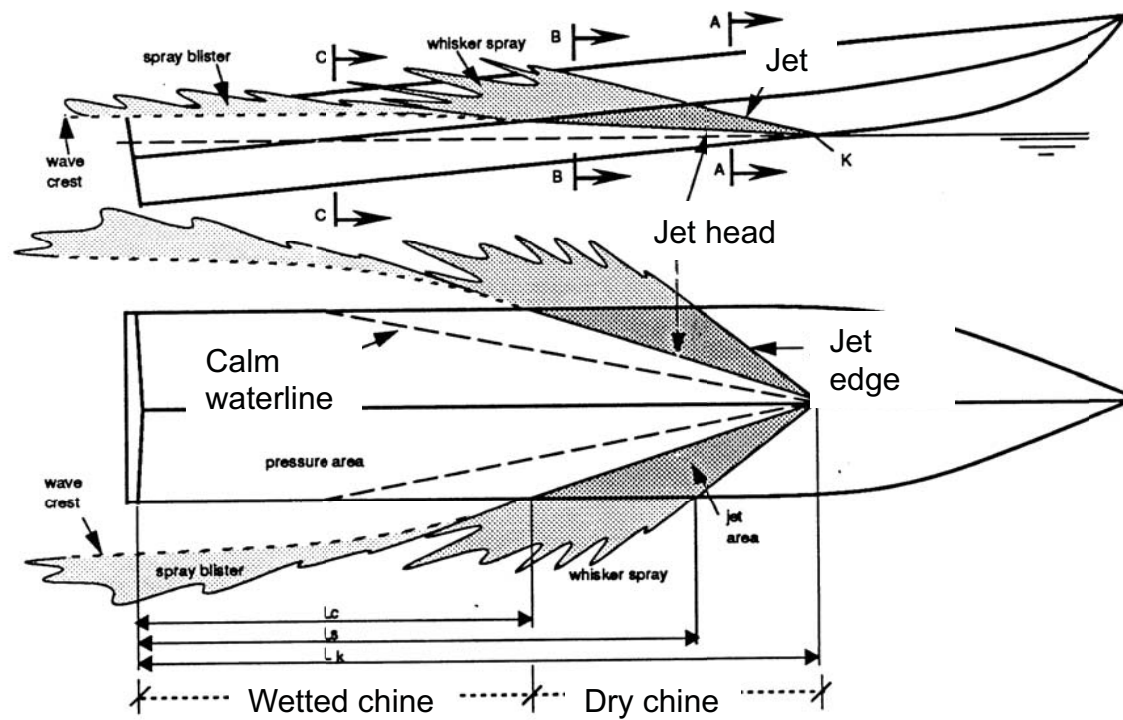
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Armin Troesch
University of Michigan
Department of Naval Architecture and Marine Engineering

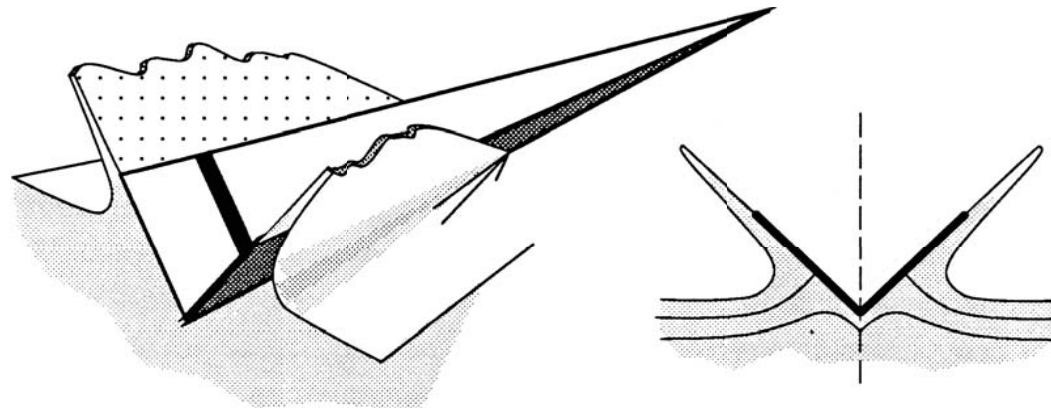
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Physics of Planing - Some Definitions



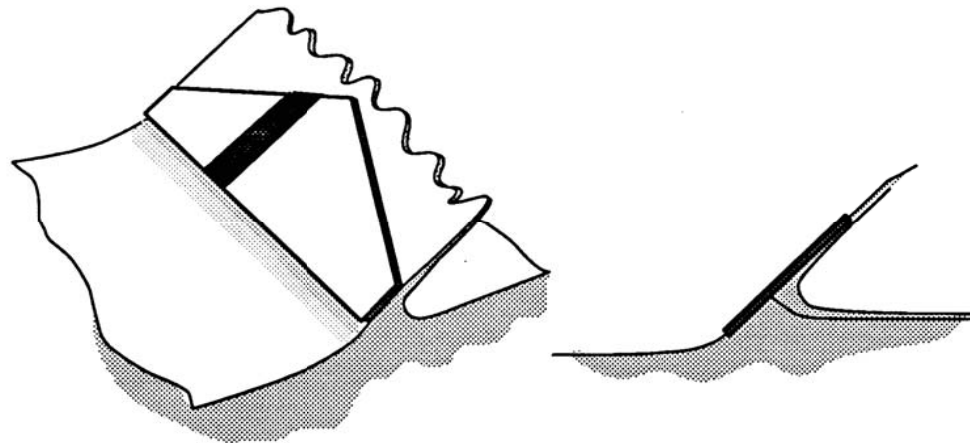
Physics of Planing - Some definitions and idealizations

Slender, low aspect ratio, prismatic hulls:
Wagner (1931, 1932), von Karman (1929)



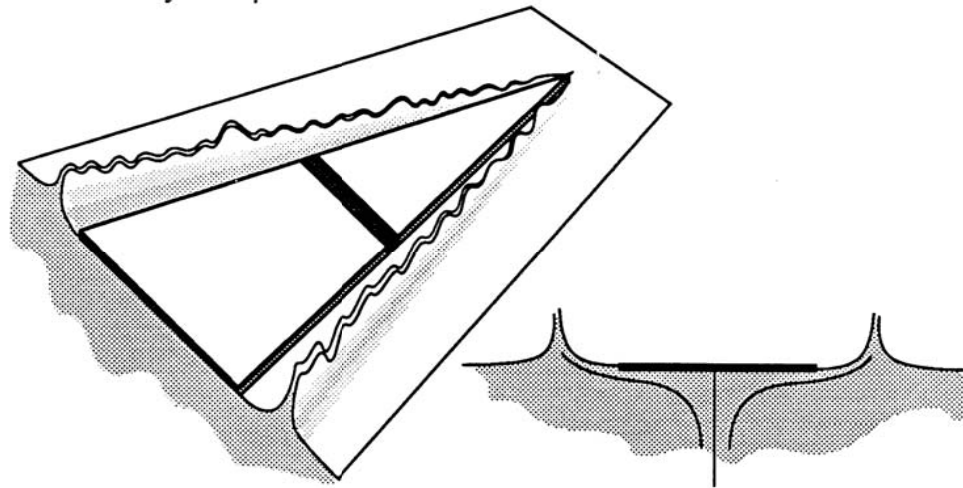
Physics of Planing - Some definitions and idealizations

High aspect ratio, flat hulls:
Green (1936)



Physics of Planing - Some definitions and idealizations

Slender, low aspect ratio, flat hulls:
Tulin (1957)

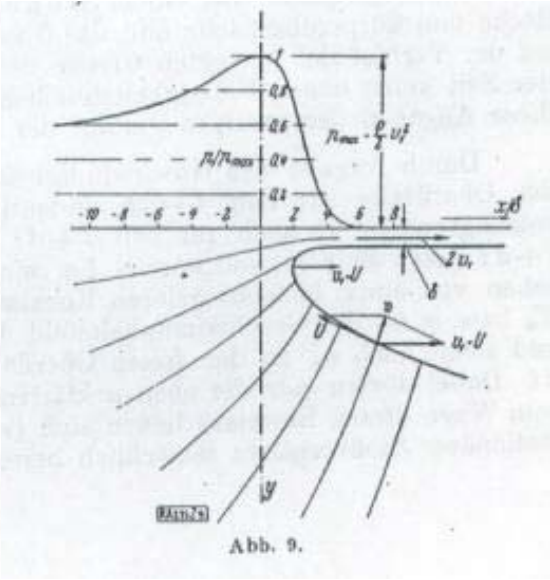
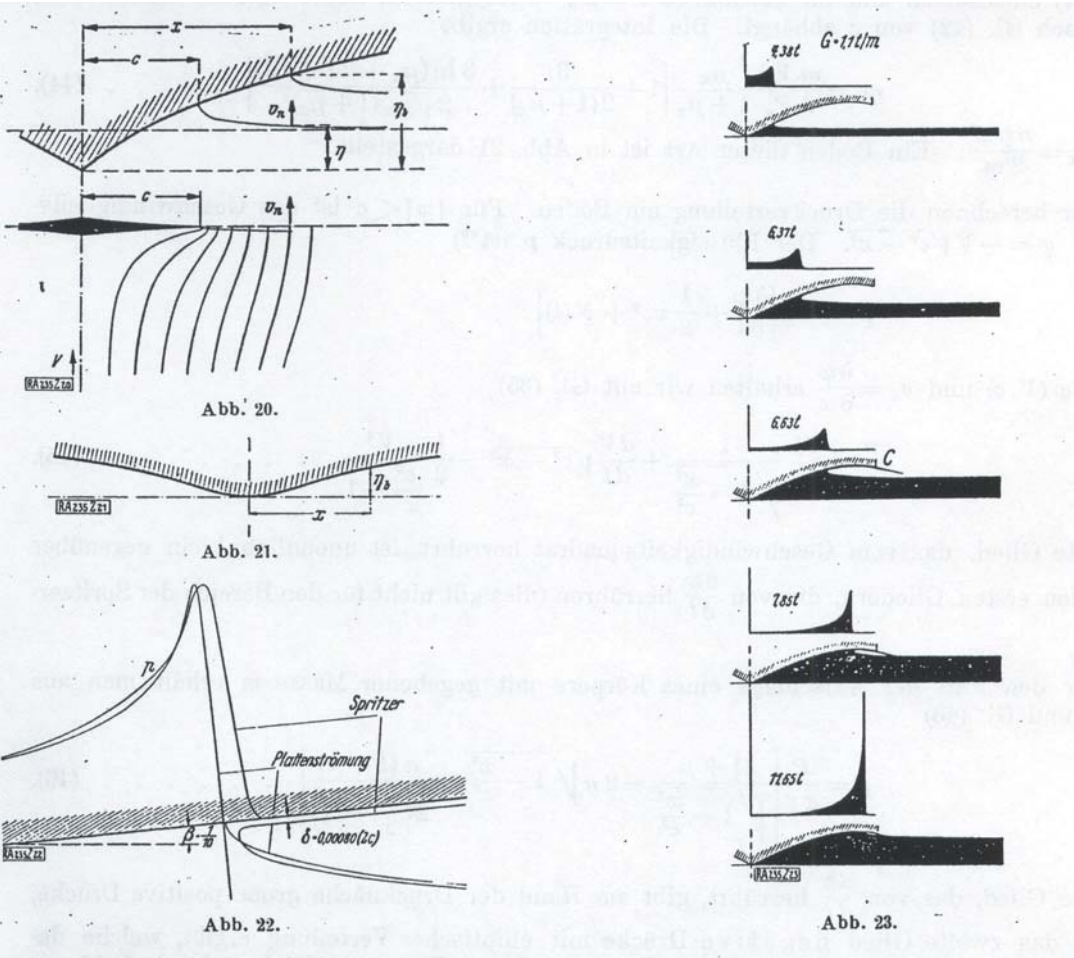


Relationship between planing and impact hydrodynamics

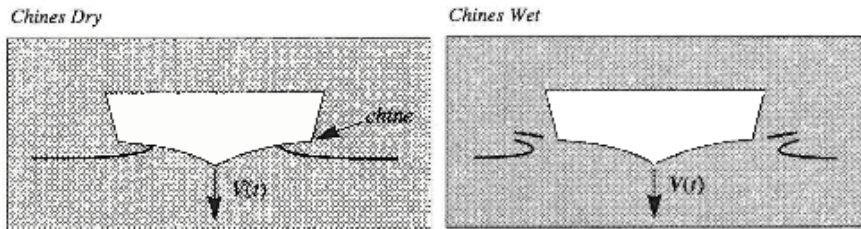
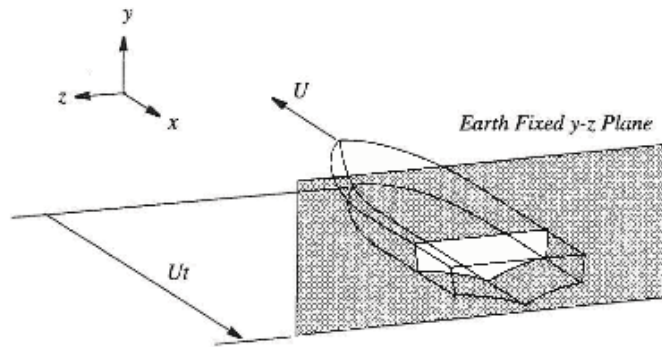
- Low order approximation
- Coordinate transformation between time and longitudinal coordinate
 $X = U t$
- Transverse plane is equivalent to 2-D impact or strip theory approximation of planing



Seminal Work in Impact Hydrodynamics, Wagner (1932) Model:

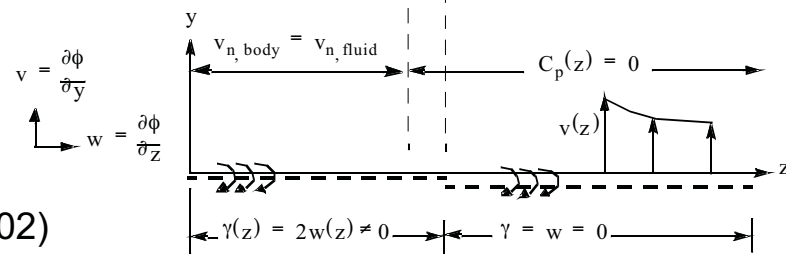
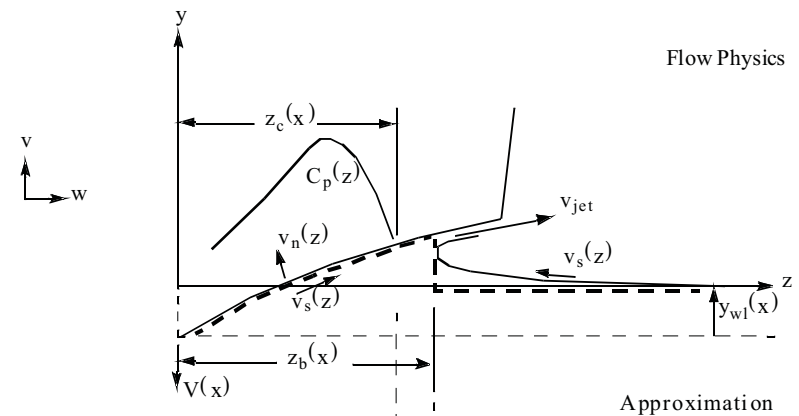
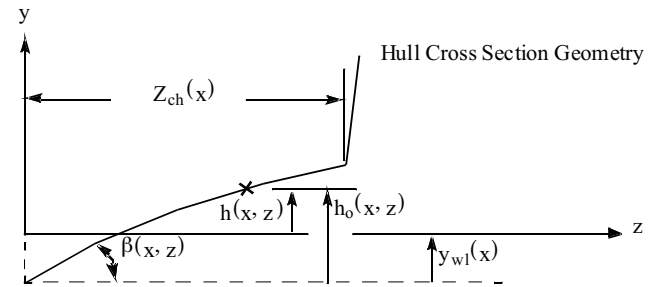


Reduction of physical model to 2-D impact solution space



Planing surface passing through earth-fixed plane

Slender Body Theory Planing Model, Savander, et al. (2002)



The Boundary Value Problem

- *Nonlinear dynamic free surface boundary condition*

$$(V_s - z_c \tau \zeta) \frac{\partial V_s}{\partial \zeta} + z_c \frac{dV_s}{d\tau} = 0 \quad 1 \leq \zeta \leq b(\tau)$$

- *Body kinematic boundary condition*

$$\gamma_c(\zeta, \tau) = - \frac{2 \cos \bar{\beta}(\zeta, \tau) \zeta \kappa(\zeta, \tau)}{\sqrt{1 - \zeta^2}} \left[V(\tau) + \frac{1}{\pi} \int_{s=1}^{b(\tau)} \frac{\gamma_s(s, \tau) ds}{\kappa(s, \tau) \sqrt{(s^2 - 1)}} + \right. \\ \left. + \frac{\zeta^2 - 1}{\pi} \int_{s=1}^{b(\tau)} \frac{\gamma_s(s, \tau) ds}{\kappa(s, \tau) \sqrt{(s^2 - 1)(s^2 - \zeta^2)}} \right] \quad 0 \leq z \leq z_c(\tau)$$



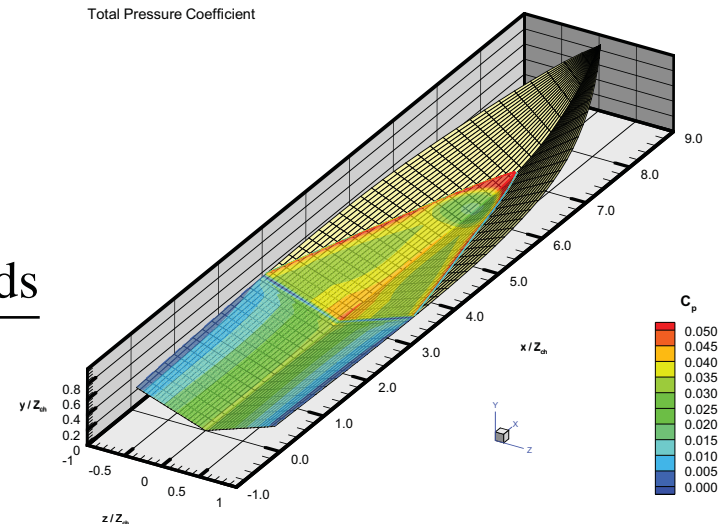
The Boundary Value Problem, con't.

- *Velocity continuity condition (i.e. Kutta condition)*

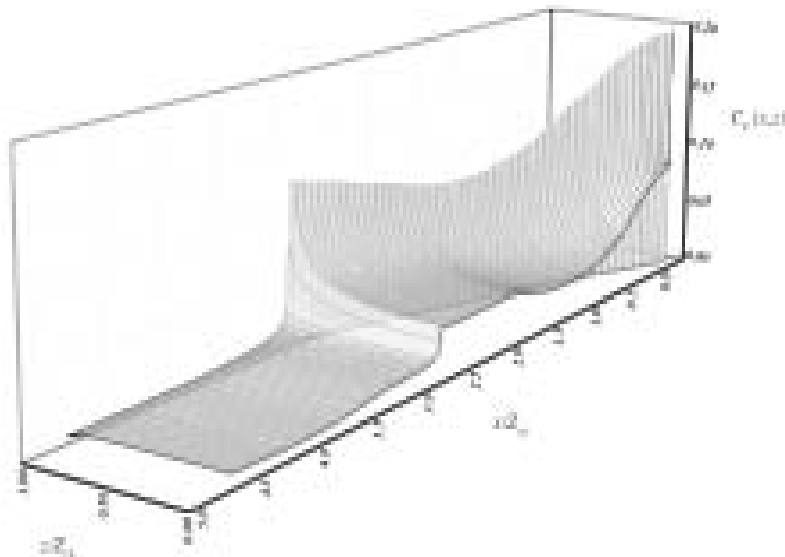
$$V(\tau) + \frac{1}{\pi} \int_{s=1}^b \frac{\gamma_s(s, \tau) ds}{\kappa(s, \tau) \sqrt{(s^2 - 1)}} = 0 \text{ at } z = z_c(z = 1)$$

- *Displacement continuity condition*

$$Y_{w1}(\tau) = \frac{2}{\pi} \int_{s=0}^1 \frac{\cos \bar{\beta}(s, \tau) h_c(s, \tau) ds}{\kappa(s, \tau) \sqrt{1 - s^2}}$$

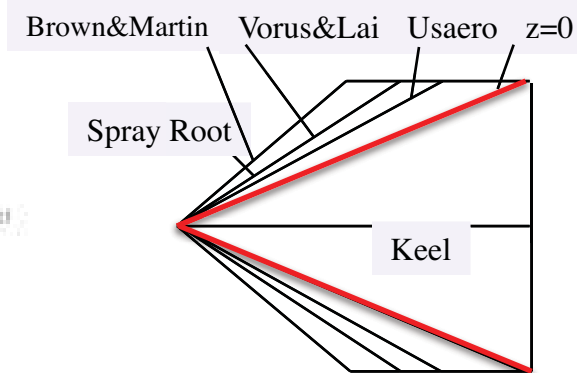


3-D Pressure Distribution and Spray Sheet

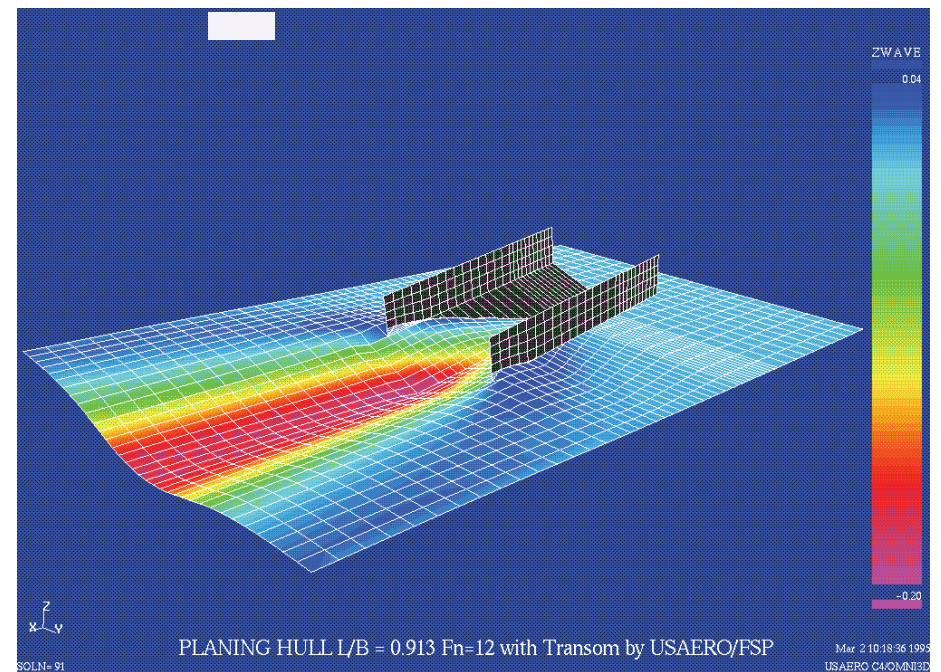


Variable deadrise 40 mph ski boat

Calculations based on 3-D boundary integral method (Savander, 1997)



| | |
|---|--------|
| C_L (Savitsky) | 0.091 |
| C_L (USAERO) | 0.053 |
| θ_0 (z=0) | 23.29° |
| θ (USAERO) | 29.17° |
| θ (Vorus&Lai) | 32.88° |
| θ (Brown&Martin) | 39.47° |
| $\lambda = 0.913, \tau = 9^\circ, \beta = 20^\circ$ | |

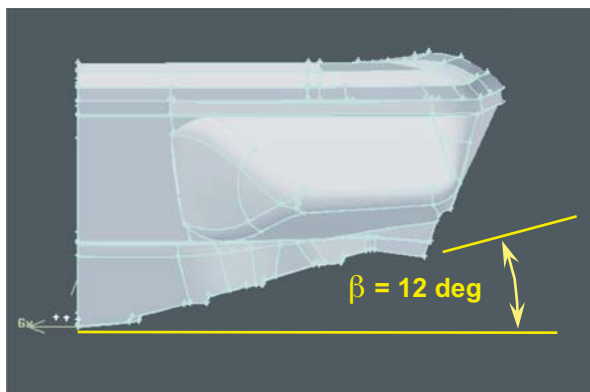


FLUENT 6.1 (CFD) Analysis: Straight Line Running

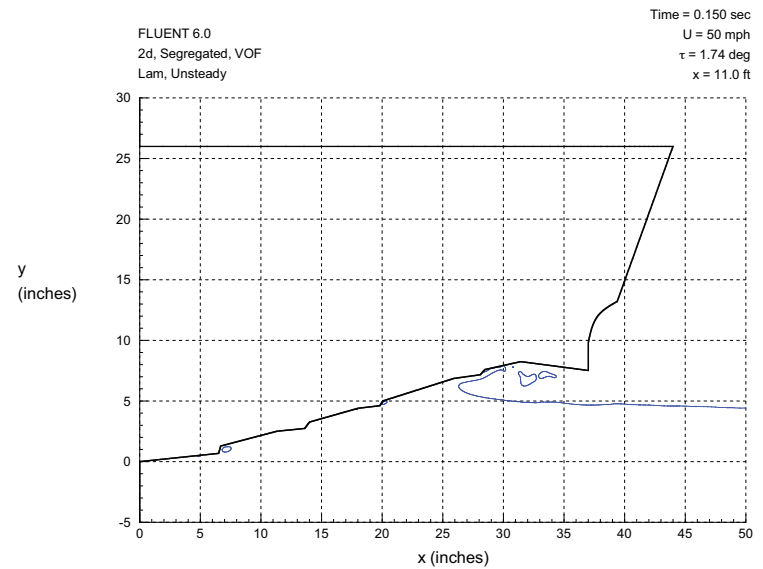
Transverse Flow @ Transom:

50 mph
 Engine Trim 1/4 Out
 Straight Line

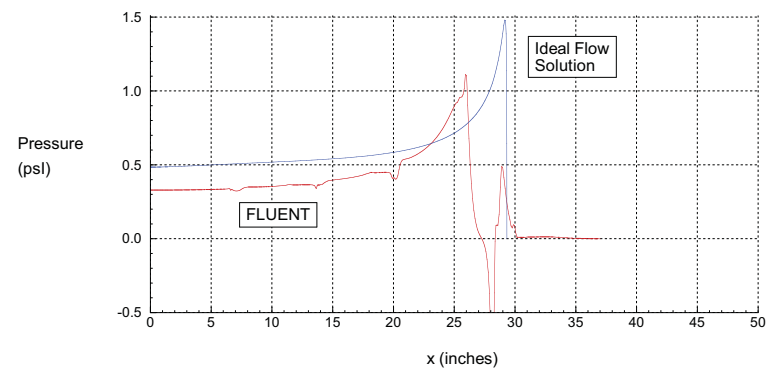
Starboard Transom of Bass Boat



Bass Boat
 Down Angle Lifting Strake Included



Section Bottom Pressure Comparison



Conclusions of planing modeling analysis

(the more significant ones?)

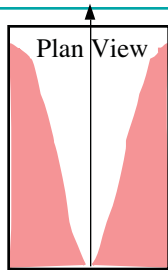
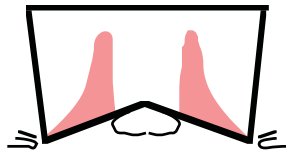
- Modeling the jet separation point (line) is critical to accurate planing predictions.
- Prediction of the correct wetted surface is critical to accurate planing predictions.
- There may be significant non-slender effects for large θ .
- The reduction to the $z=0$ plane works well in the chines dry regime but less so in the chines wet regime.
- Both slender body theory and 3-D Models have the potential to predict steady lift and drag accurately enough for design purposes.



Validation and Application of Technology



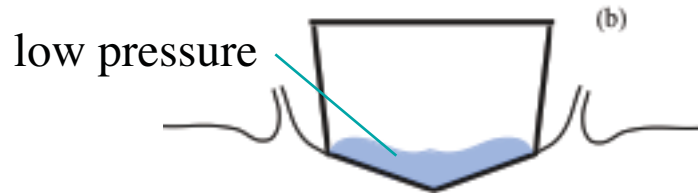
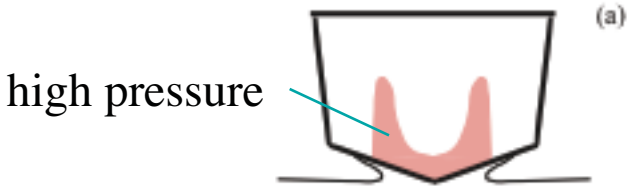
Calculated variation of deadrise to achieve maximum lift to drag ratio



Extension of 2-D Planing Theory to Innovative Hull Form Design

(a) High pressure region in chines dry impact

(b) Low pressure region in chines wet impact



Validation and Application of Technology

SOLAR SPLASH: Intercollegiate Solar/Electric Boat Regatta

featuring

The University of Michigan's Inverted "V" Planing Hull



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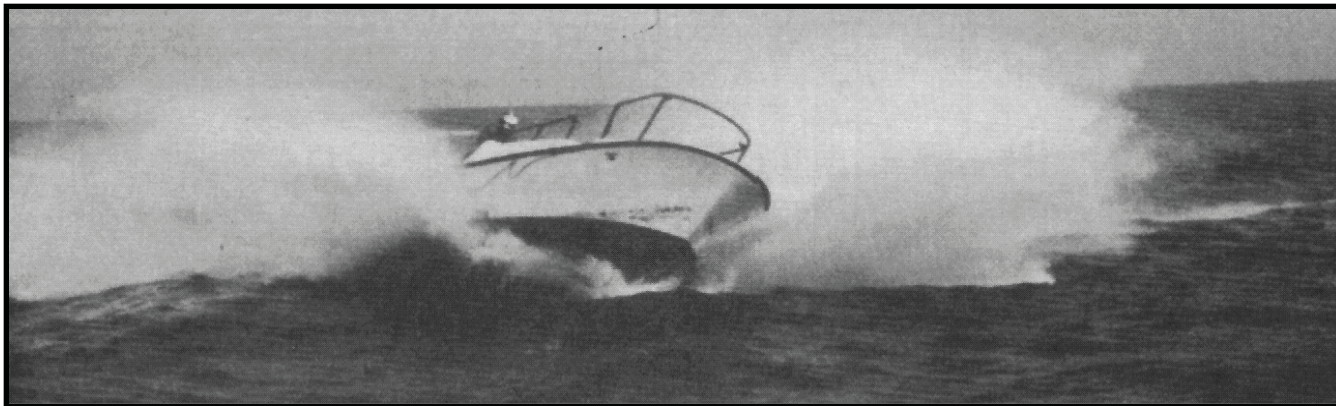
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Selected experiments and theories
with practical applications
- **Nonlinear Dynamics Analysis Applied to Planing Hulls**
- Stochastic Vibro-Impact Model for Extreme Planing
Craft Acceleration Estimation



RATIONALE FOR DYNAMIC ANALYSIS:

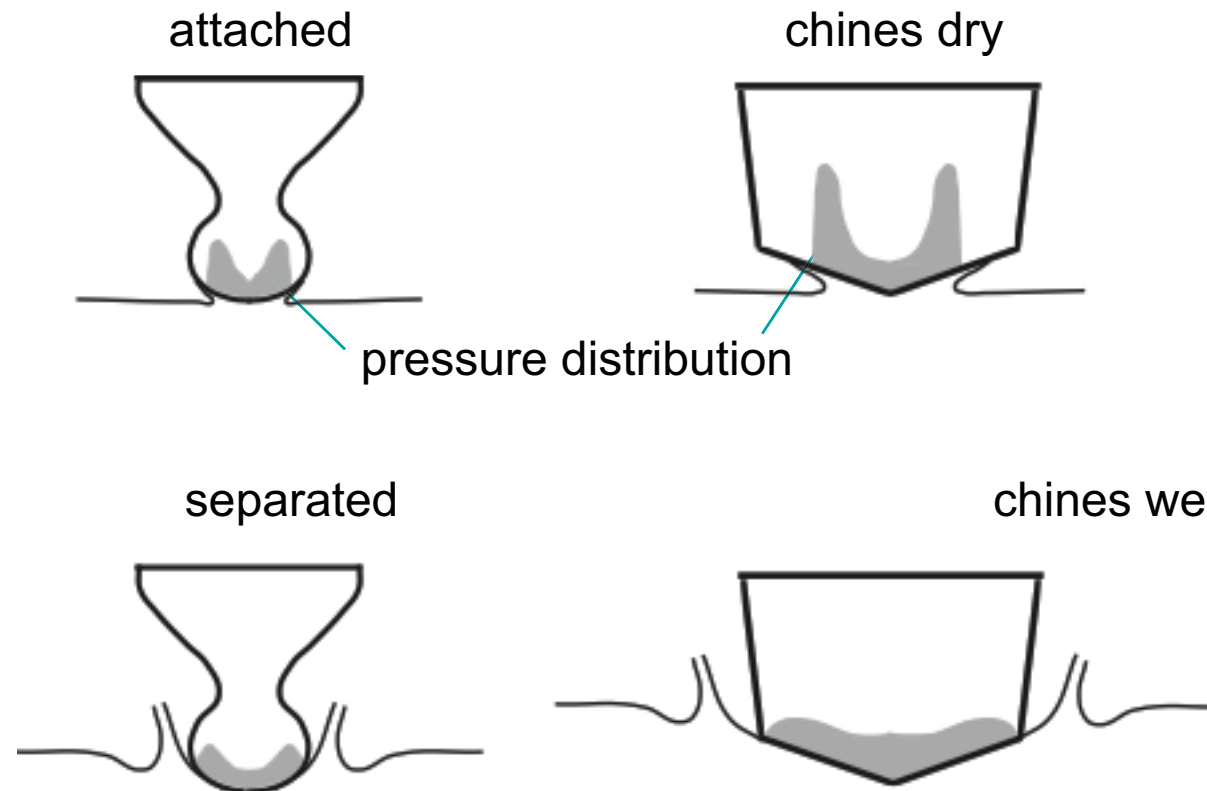
Identification of design parameters that are critical to performance.



Example: “A Case Study of Dynamic Instability in a Planing Hull”, Codega and Lewis. *Marine Tech.*, April, 1987



Problem Definition : Steady Symmetric Planing



Problem Definition : Steady Asymmetric Planing - Type "A"

attached



chines dry



separated



chines dry/wet

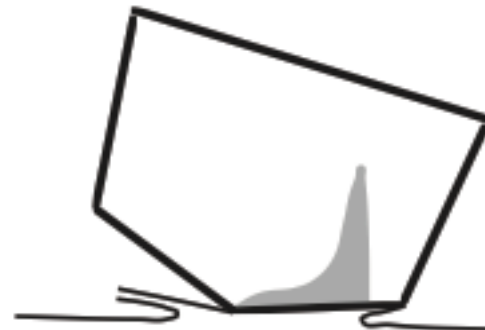


Problem Definition : Steady Asymmetric Planing - Type "B"

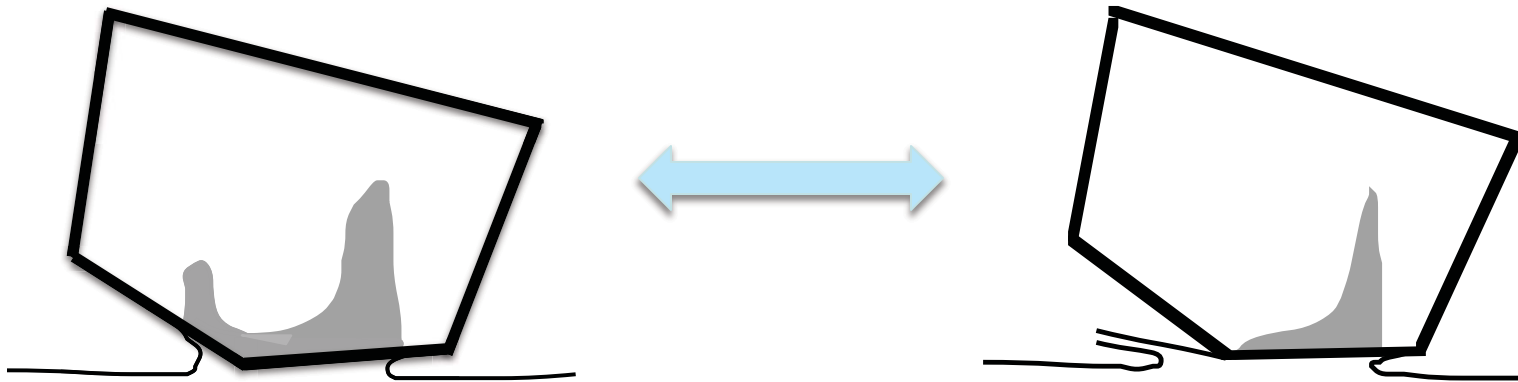
keel separation/attached



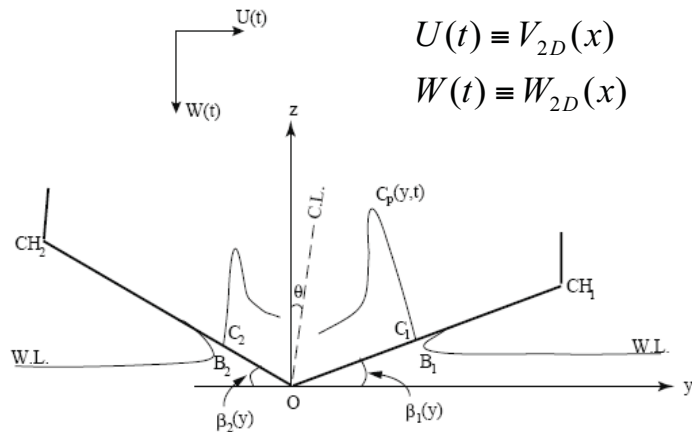
keel separation/chines dry



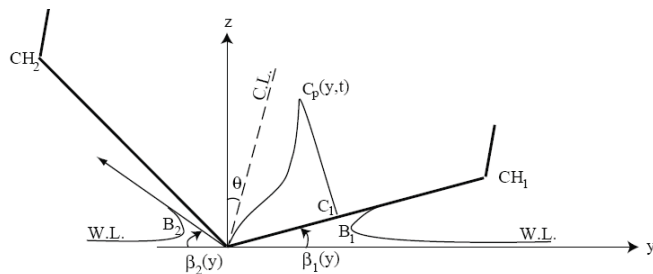
Transition to Steady Asymmetric Planing - Type "B"



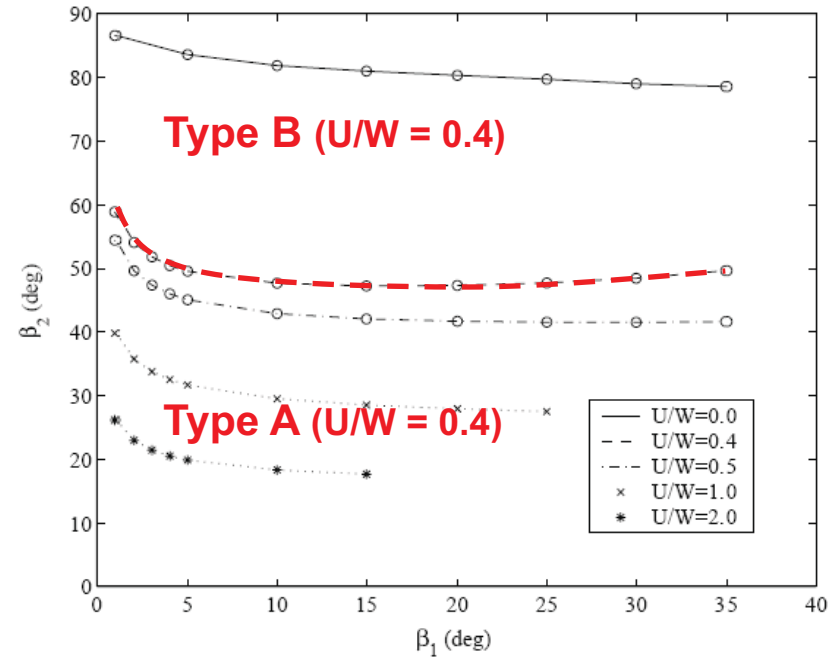
Steady Transverse Planing Modeling: Asymmetric Impact Theory



Type A model of cylinder asymmetric impact (small asymmetry) and horizontal impact velocity



Type B model of cylinder asymmetric impact (large asymmetry) and horizontal impact velocity



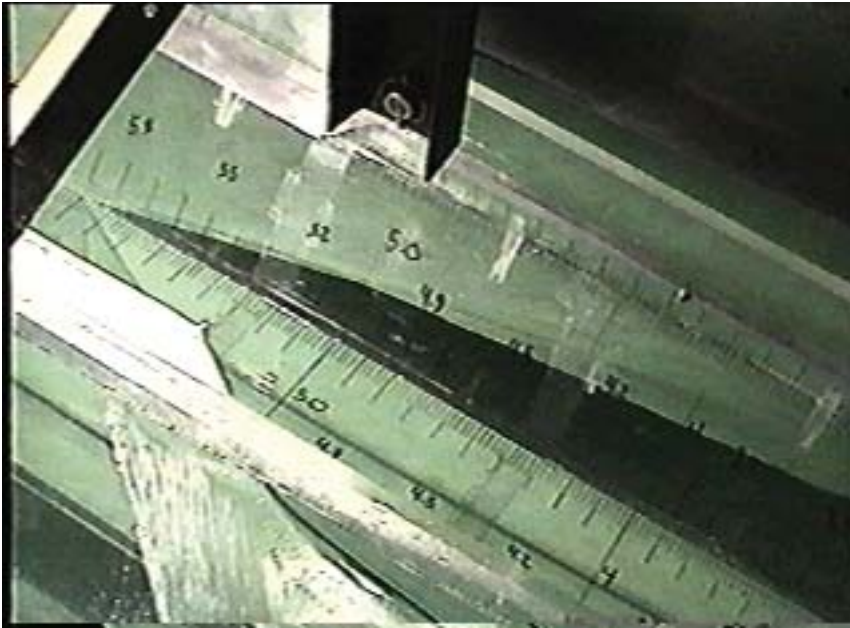
Critical angle β_2 versus the corresponding β_1 at which ventilation occurs off the keel for different ratios of impact velocities

(Judge et al, 1999
Xu et al, 1998, 1999)

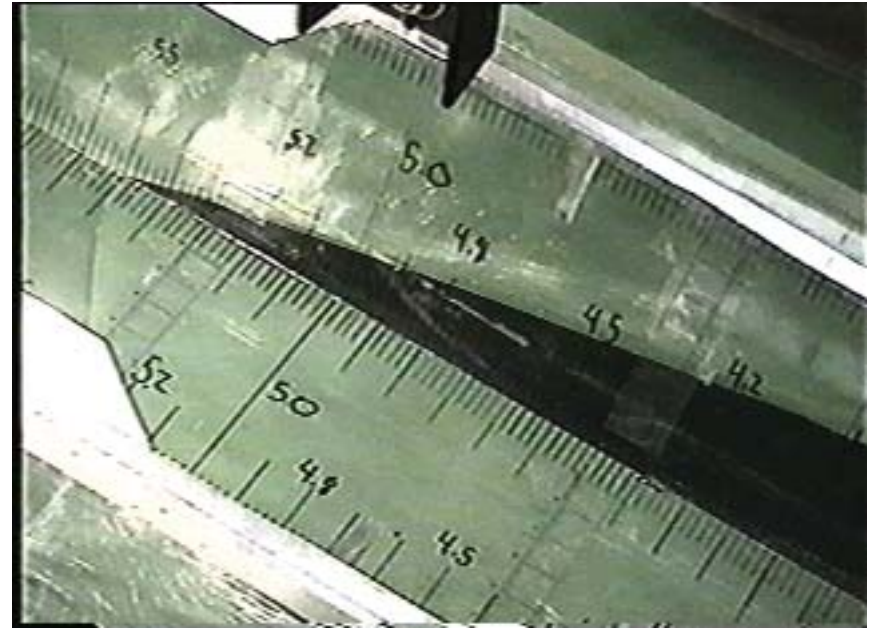


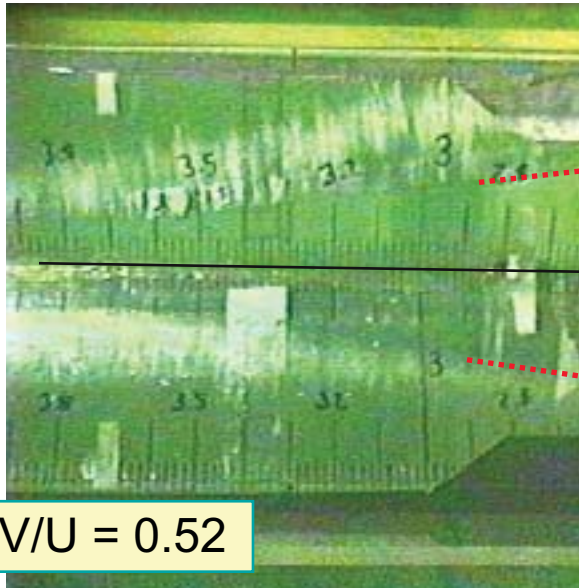
Steady Transition to asymmetric planing - Type "B"

35 deg deadrise heeled 20 deg



35 deg deadrise heeled 29 deg



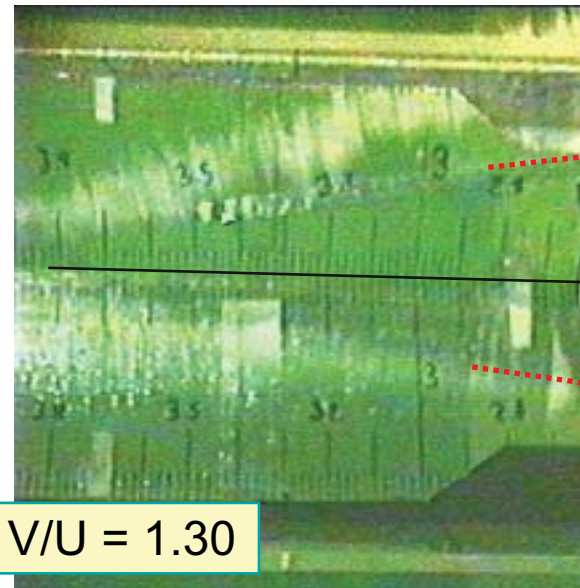


$V/U = 0.52$

Jet Root

Keel

Jet Root

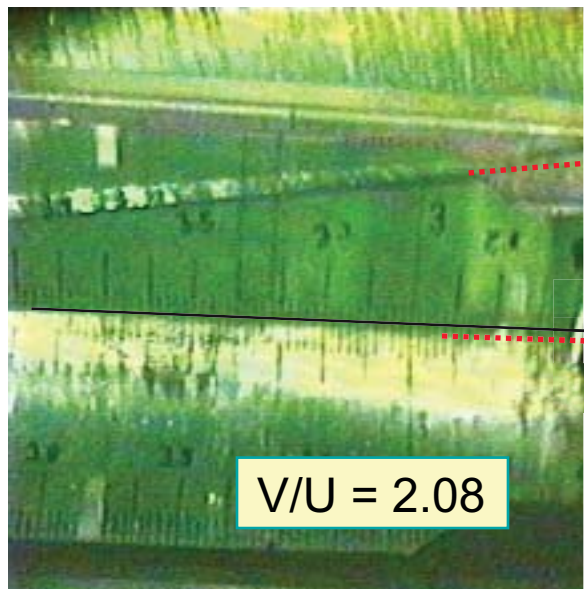


$V/U = 1.30$

Jet Root

Keel

Jet Root




$V/U = 2.08$

Jet Root

Keel

Jet Root

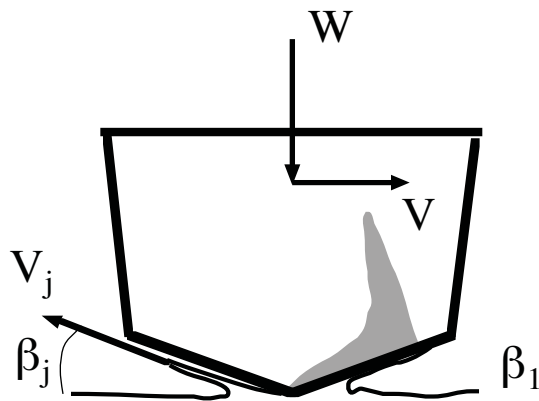
Onset of Separation
Due to Transverse
Velocity

Direction of V 



Steady Asymmetric planing due to horizontal velocity

Estimate of the horizontal velocity required for separation:



$$\frac{V}{W} \cong \frac{V_j/W \sin\beta_j - \tan\beta_1 \cos\beta_j}{\tan\beta_1}$$

| β_1 | V/W |
|------------|---------|
| 20° | 4.2-4.5 |
| 35° | 2.0-2.5 |

Ref: Xu, Troesch, 1999 and Judge, et al. 2002



The Synergy of Experiments and Theory

Experiments with prismatic hull forms

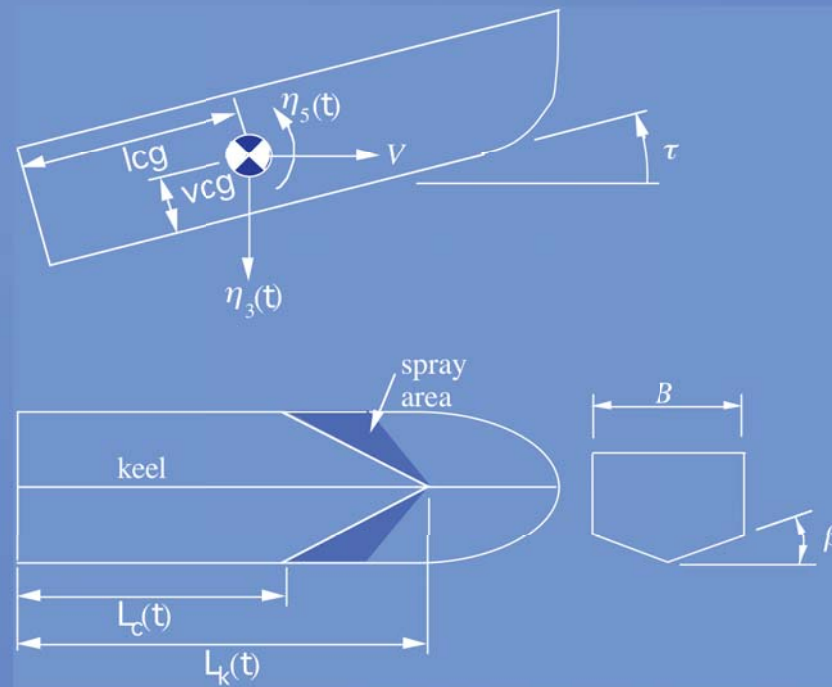
- Steady planing
- Unsteady planing - vertical plane motions

For example: Savitsky (1964), Altman (1968),
Fridsma (1969, 1971), De Zwaan (1976), Savitsky &
Brown (1976), Troesch (1992), and others.....



On the Hydrodynamics of Vertically Oscillating Planing Hulls

HULL and WETTED LENGTH DEFINITIONS



Equations of Motion

$$Z = m \ddot{\eta}_3(t)$$

$$M = I_{55} \ddot{\eta}_5(t)$$

May be **Modeled** as

$$[m + \mathbf{A}(\eta; \omega)]\{\ddot{\eta}(t)\} + [\mathbf{B}(\eta; \omega)]\{\dot{\eta}(t)\} + [\mathbf{C}(\eta)]\{\eta(t)\} = \{F(t)\}$$



Geometry: Dynamic wetted length

$$L(t) = lcg + \frac{vcg}{\tan(\tau - \eta_5(t))} - \frac{(z_{wl} + \eta_3(t))}{\sin(\tau - \eta_5(t))}$$

$$\lambda_{ave} = (\lambda_k + \lambda_c)/2 + 0.03$$

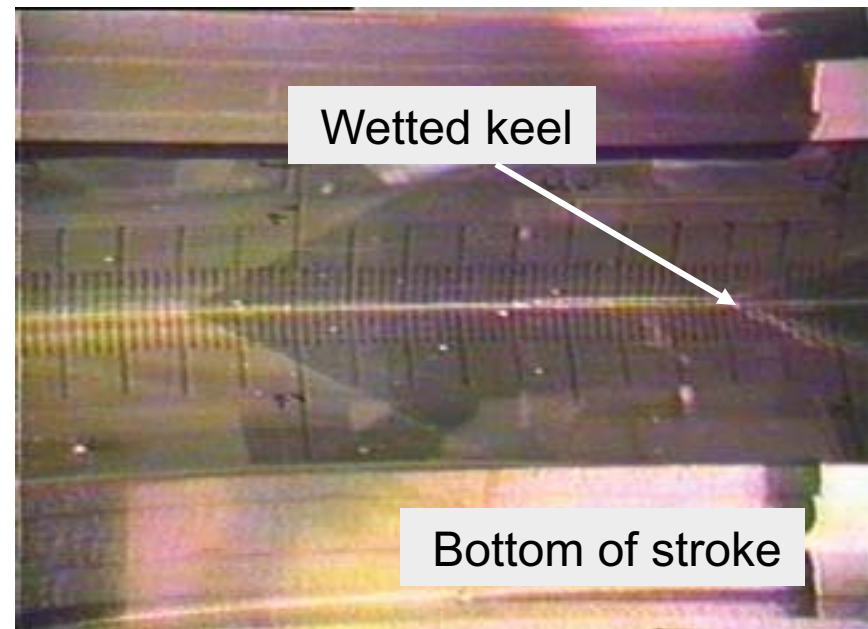
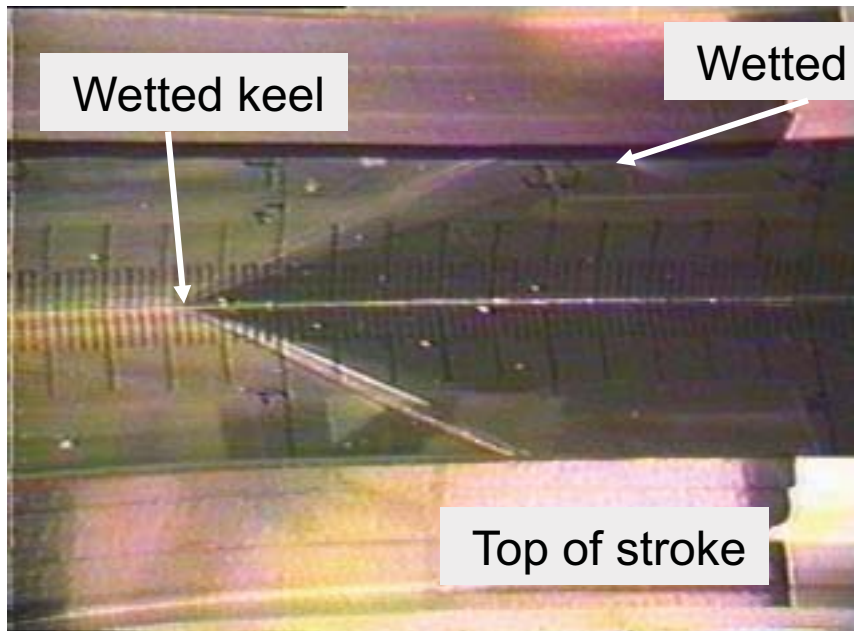
$$\hat{\lambda}_k = \hat{\lambda}_{ave} + 0.5(0.57 + 0.001\beta)(\tan \beta / (2 \tan(\tau - \eta_5)) - 0.006\beta) - 0.03$$

$$\hat{\lambda}_c = \hat{\lambda}_k - (0.57 + 0.001\beta)(\tan \beta / (2 \tan(\tau - \eta_5)) - 0.006\beta) \quad \text{for } \hat{\lambda}_c \geq 1$$

Steady mean wetted length from Savitsky (1964) and Savitsky & Brown (1976) modified for motions (Troesch 1992)



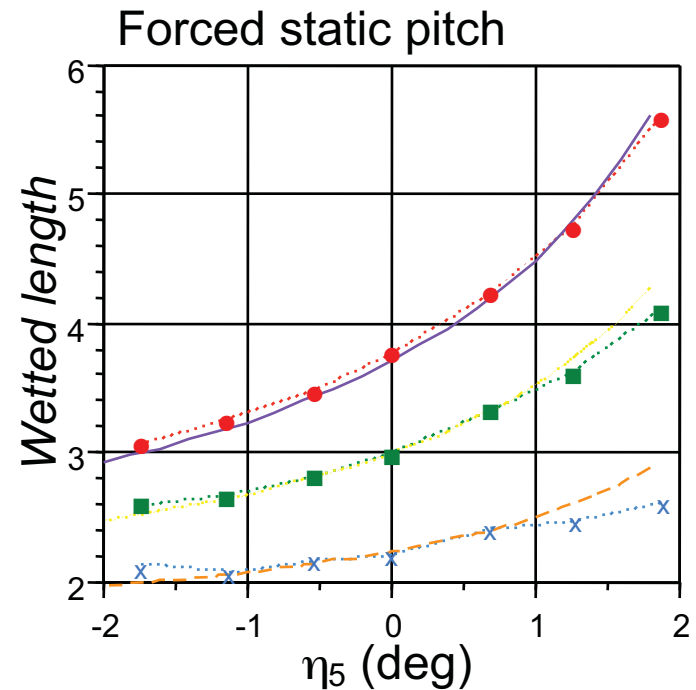
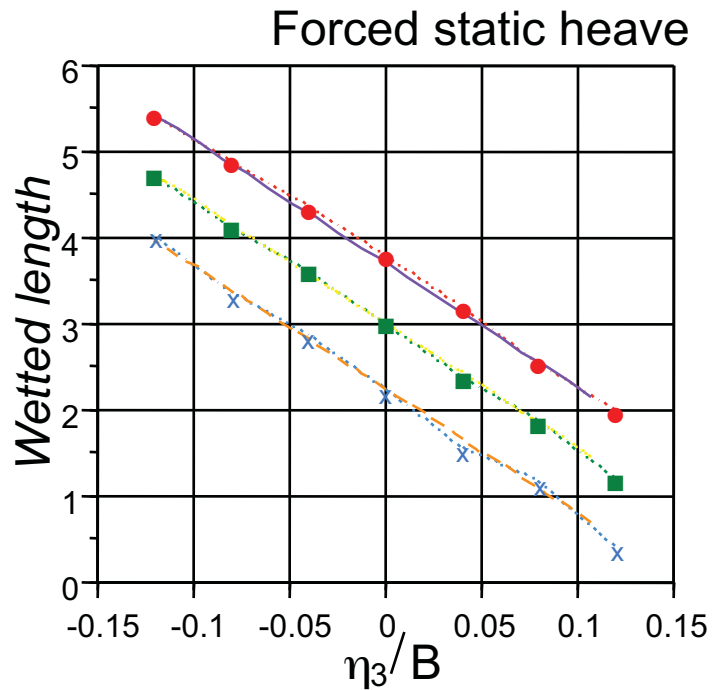
Geometry: Dynamic wetted length in heave



Geometry: Dynamic wetted length in heave



Geometry: Dynamic wetted length



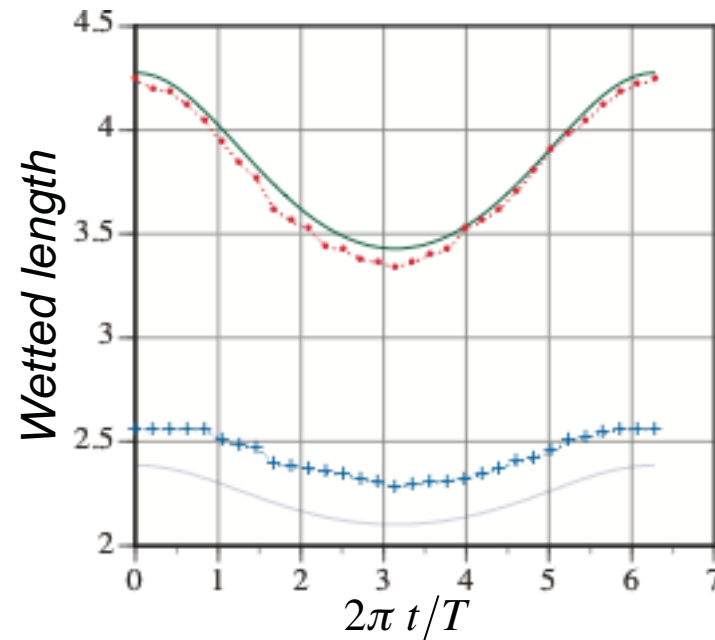
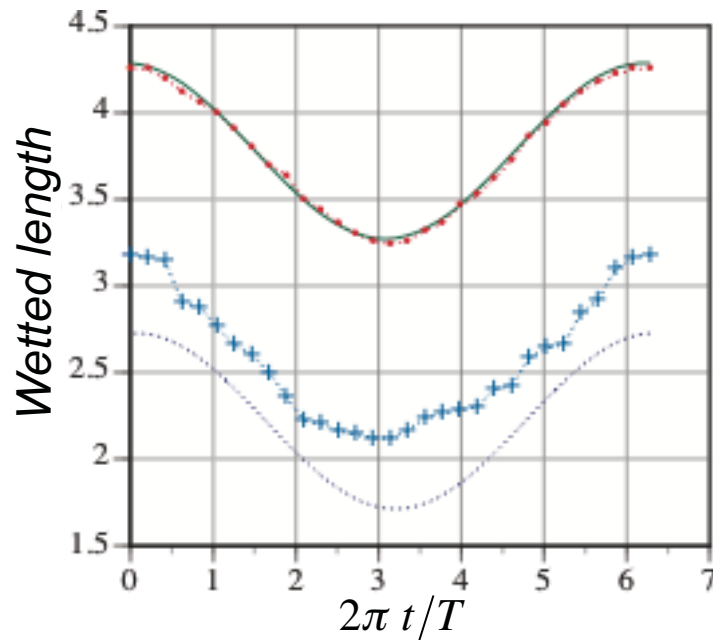
Wetted length versus constant heave and pitch displacement. \blacksquare -, λ_{ave} ;
 ---, Eq. (5); \bullet -, λ_K ; —, Eq. (6); \times -, λ_C ; - - -, Eq. (7).



Geometry Dynamic wetted length

Forced heave

Forced pitch

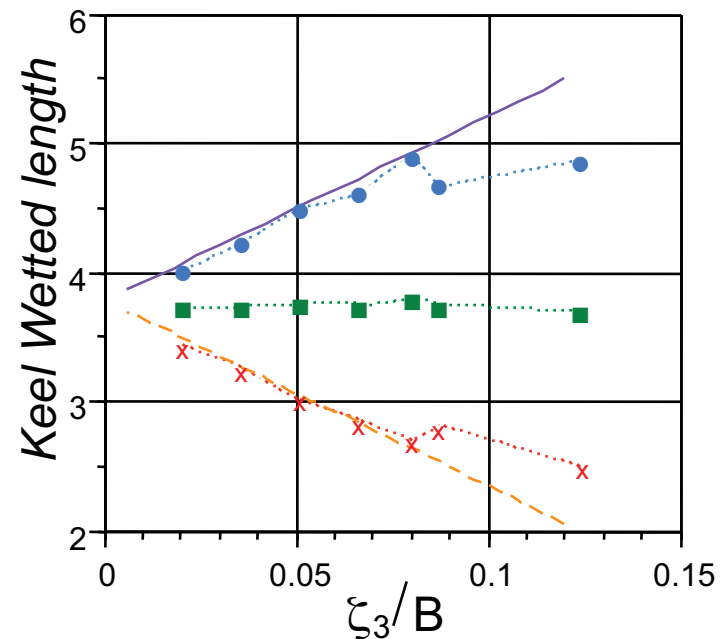
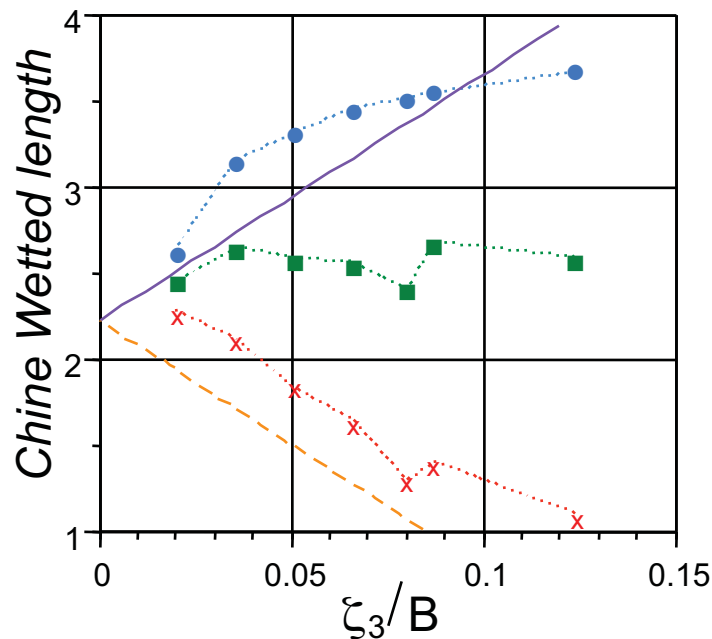


Wetted length during one cycle of heave and pitch oscillation.
 $\xi_3/B = 0.036$, $\omega\sqrt{B/g} = 1.13$. + , λ_c ; - - -, Eq. (4); ● , λ_k ; —, Eq. (4).



Geometry: Dynamic wetted length

Forced heave

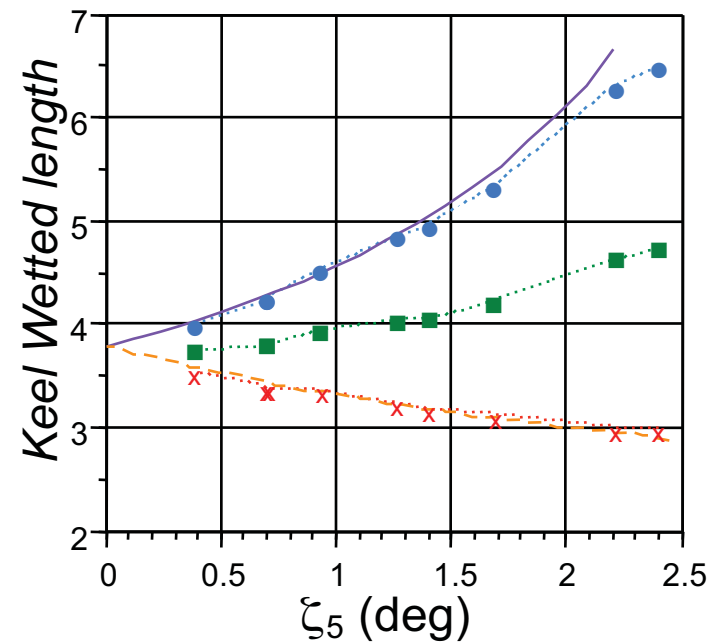
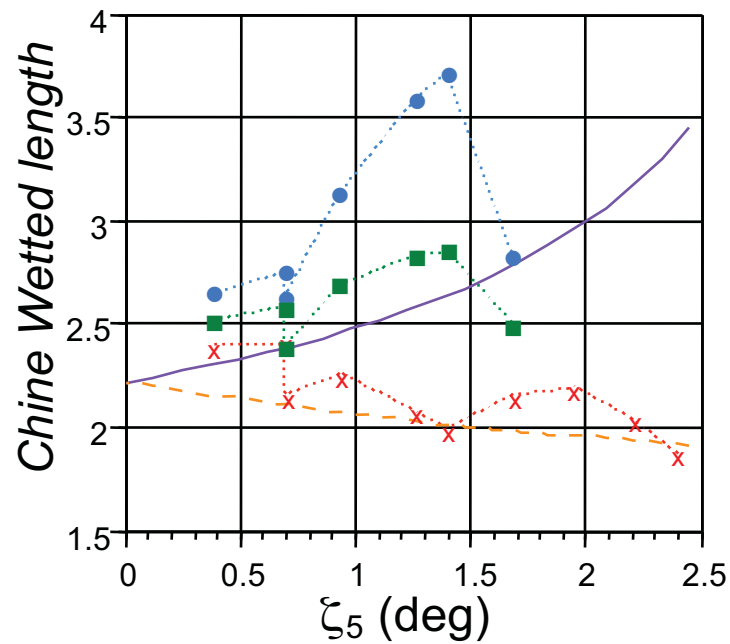


Wetted maximum and minimum chine and keel lengths as a function of heave amplitude. \bullet , λ_{max} ; —, Eq. (4); \times , λ_{min} ; - - -, Eq. (4); \blacksquare , λ_{ave}



Geometry: Dynamic wetted length

Forced pitch

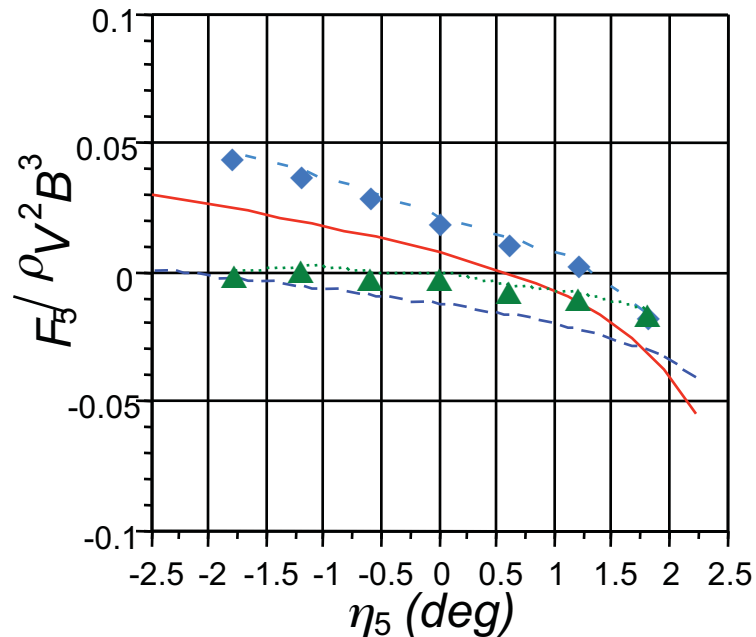


Wetted maximum and minimum chine and keel lengths as a function of pitch amplitude. \bullet , λ_{max} ; —, Eq. (4); \times , λ_{min} ; - - -, Eq. (4); \blacksquare , λ_{ave} .

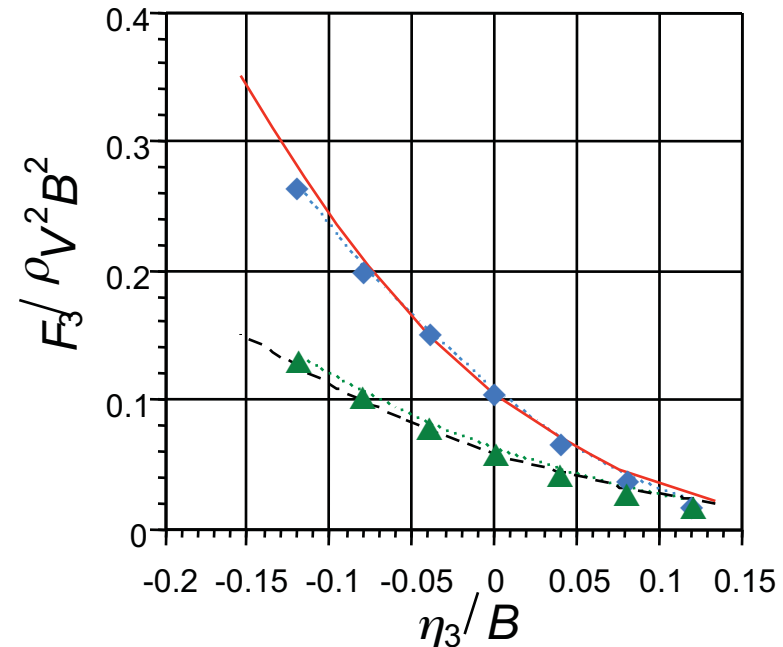


Hydrodynamic stiffness, C_{55} and C_{33}

Forced pitch



Forced heave

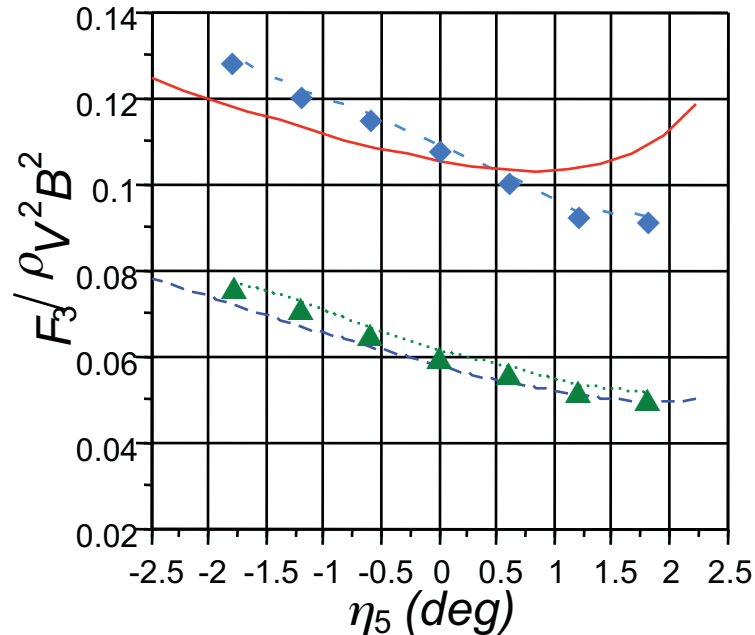


: Total vertical force and moment versus constant linear heave and pitch displacement. $L/B = 3.0$, $\tau = 4.0$ deg. $C_V = 1.5$; $-\diamond-$, exp; $—$; Savitsky [14]. $C_V = 2.5$; $-\blacktriangle-$, exp; $- - -$, Savitsky [14].

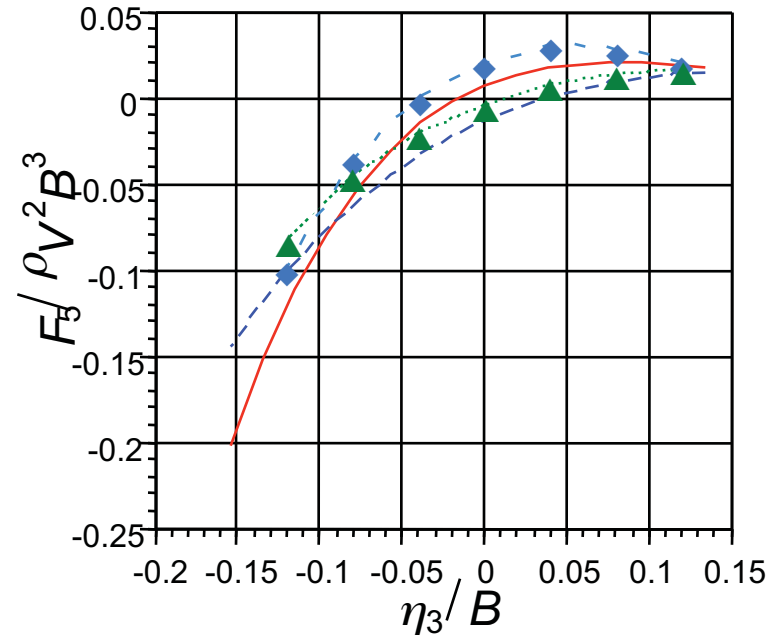


Hydrodynamic stiffness, C_{35} and C_{53}

Forced pitch



Forced heave



: Total vertical force and moment versus constant linear heave and pitch displacement. $L/B = 3.0$, $\tau = 4.0$ deg. $C_V = 1.5$; -◆-, exp; —, Savitsky [14]. $C_V = 2.5$; -▲-, exp; - - -, Savitsky [14].



Added mass and damping, A_{33} and B_{33}

$$-\omega^2(m + A_{33}) + i\omega B_{33} + C_{33} = (f_1 + f_2)/\zeta_3 \quad \Rightarrow \quad \sum F = m\ddot{\zeta}_3$$

$$-\omega^2(-mx_{cg} + A_{53}) + i\omega B_{53} + C_{53} = (f_1 - f_2)L/\zeta_3 \quad \Rightarrow \quad \sum M = 0$$

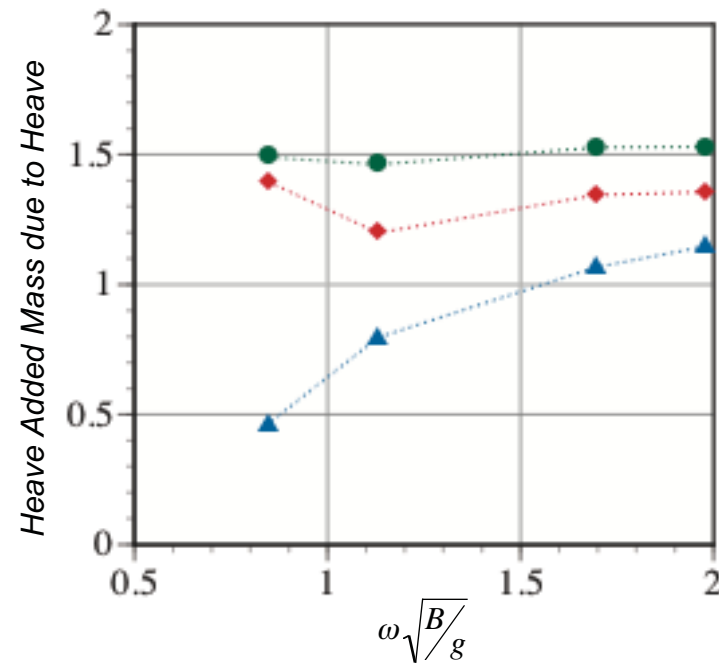
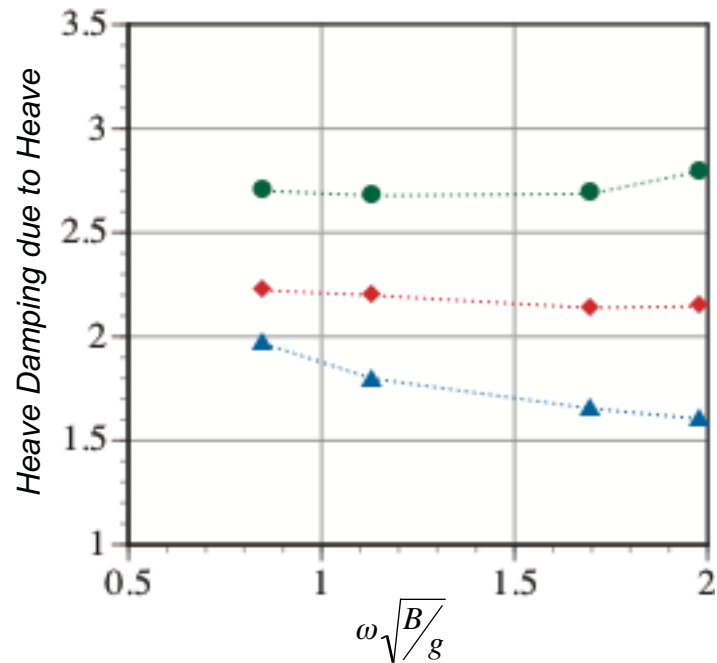
Which reduce to

$$A_{33} = \frac{1}{-\omega^2} \left[\frac{\text{mag}(f_1 + f_2)}{\text{mag}(\zeta_3)} \cos(\arg(f_1 + f_2) - \arg(\zeta_3)) - C_{33} \right] - m$$

$$B_{33} = \frac{1}{\omega} \left[\frac{\text{mag}(f_1 + f_2)}{\text{mag}(\zeta_3)} \sin(\arg(f_1 + f_2) - \arg(\zeta_3)) \right]$$



Added mass and damping, A_{33} and B_{33}



Added mass and damping coefficients in heave versus frequency $L/B = 4.0$;
 $\tau = 6$ deg. -▲-, $C_v = 1.5$; -◆-, $C_v = 2.0$; -●-, $C_v = 2.5$.



Conclusions of forced motion experiments

Wetted length: The wetted length and equivalently the wetted surface of an oscillating planing hull is strongly time dependent. For low to moderate amplitudes of motion, the keel wetted length can be treated kinematically as the intersection between the keel and a stationary free surface. The chine wetted length is influenced by spray jet dynamics.



Conclusions of forced motion experiments (con't.)

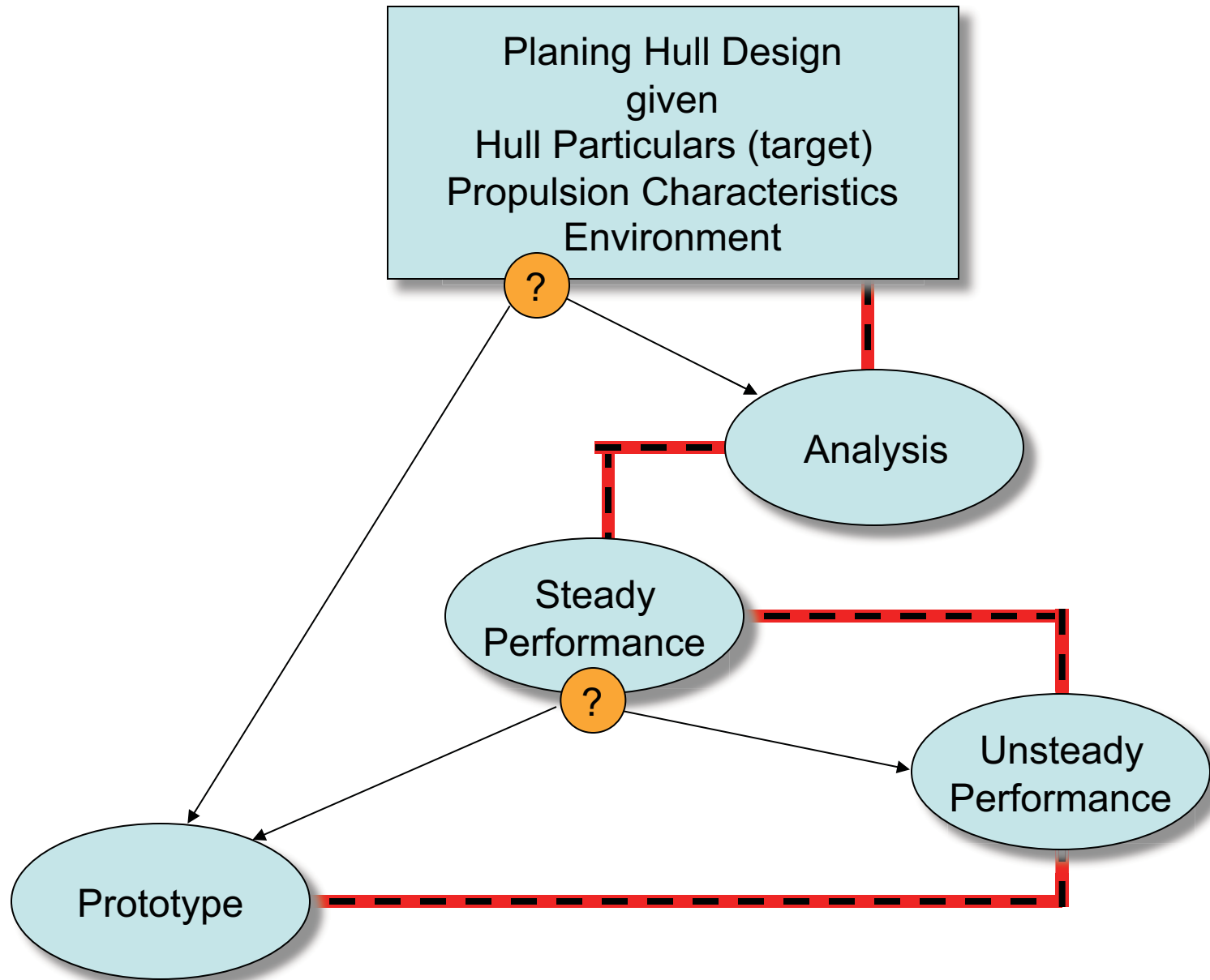
Restoring force matrix, C . The restoring force matrix is amplitude dependent. A good approximation for prismatic hulls can be found by applying the results of Savitsky [14] with the mean wetted, λ , and effective trim expressed as functions of the heave and pitch motions.



Conclusions of forced motion experiments (con't.)

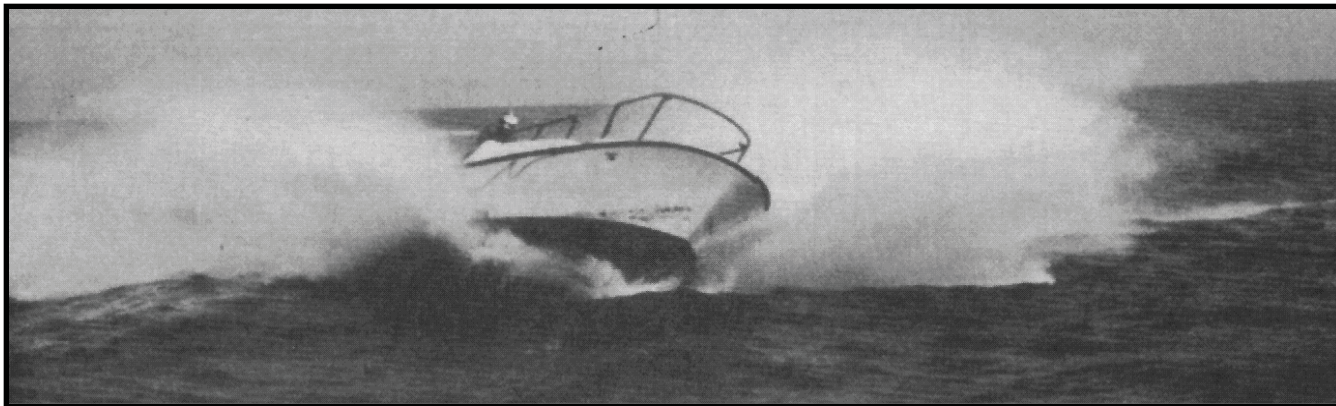
Added mass and damping matrices, A and B . For low to moderate planing speeds, $C_V=1.5 - 2.5$, the added mass coefficient, and to a lesser extent the damping coefficient, can be frequency dependent. When the experimental results were analyzed using a constant restoring force matrix, the added mass and damping in heave, A_{j3} and B_{j3} , $j=3,5$ showed relatively little amplitude dependency compared to the added mass and damping in pitch, A_{j5} and B_{j5} , $j=3,5$.





RATIONALE FOR DYNAMIC ANALYSIS:

Identification of design parameters that are critical to performance.



Example: “A Case Study of Dynamic Instability in a Planing Hull”, Codega and Lewis. *Marine Tech.*, April, 1987



THE ROLE OF SIMULATION IN DESIGN

With regards to the new computer simulators... "If you choose the right parameters, your simulated experiment will normally work just like the corresponding real-world experiment. (However,) there are some edges to this simulated world, and if you step over the simulation breaks down badly..." (Swaine, 1992)



Analysis and Simulation of Nonlinear Planing Hull Dynamics

- **General description** of methods of modern nonlinear systems analysis
- Examples of analysis
 - Ref: Troesch and Falzarano (1993),
Troesch and Hicks (1994)
Hicks, Troesch and Jiang (1995)



Analysis and Simulation of Nonlinear Planing Hull Dynamics ***Simple Model*** - Complex Dynamics

Ref: Savitsky (1964)
Guckenheimer and Holmes (1983)
Thompson and Stewart (1986)
Seydel (1988)
Troesch (1992)
Troesch and Falzarano (1993)



Analysis and Simulation of Nonlinear Planing Hull Dynamics **Complex Model** - Complex Dynamics

Ref: Zarnick (1978)
Guckenheimer and Holmes (1983)
Seydel (1988)
Akers et al. (1999, etc.)
Hicks et al. (1994, 1995)



VERTICAL PLANE EQUATIONS OF MOTION

Linear (unforced) (Martin, 1978)

$$[\mathbf{A}]\{\ddot{\eta}(t)\} + [\mathbf{B}]\{\dot{\eta}(t)\} + [\mathbf{C}]\{\eta(t)\} = \{0\}$$

Equivalent nonlinear model (unforced) (Hicks et al, 1994)

$$[\mathbf{A}]\{\ddot{\eta}(t)\} + [\mathbf{B}]\{\dot{\eta}(t)\} + [\overline{\mathbf{C}}(\eta(t))]\{\eta(t)\} = \{0\}$$

Equivalent nonlinear model (forced) (Hicks et al, 1994)

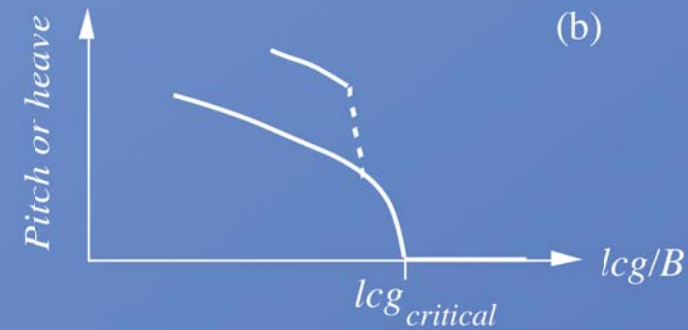
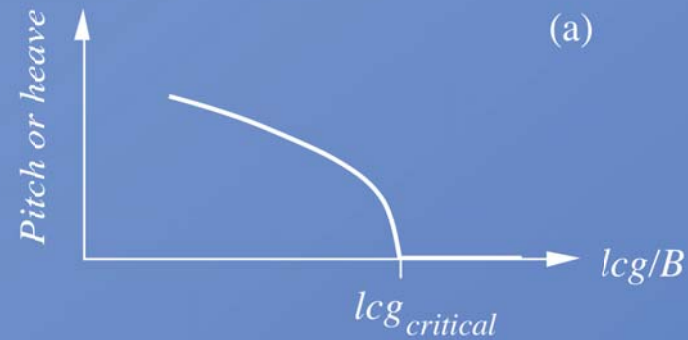
$$[\mathbf{A}]\{\ddot{\eta}(t)\} + [\mathbf{B}]\{\dot{\eta}(t)\} + [\overline{\mathbf{C}}(\eta(t))]\{\eta(t)\} = \{F(t)\}$$

Fully nonlinear simulation (forced) (Zarnick, 1978, Akers, 1994)

$$[\mathbf{M}]\{\ddot{\eta}(t)\} = \{F(t)\}$$

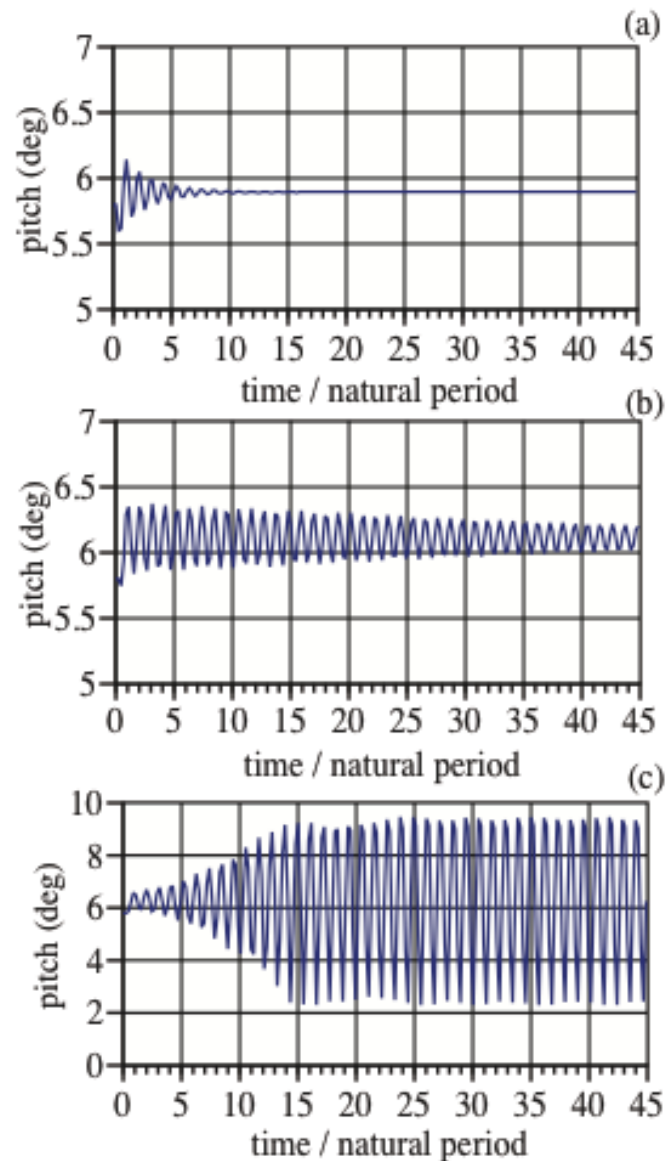


SCHEMATIC OF HOPF BIFURCATIONS (Unforced motion, i.e. porpoising)



(a) Single branch (b) Multiple branches





SIMULATION: TRANSITION TO PORPOISING

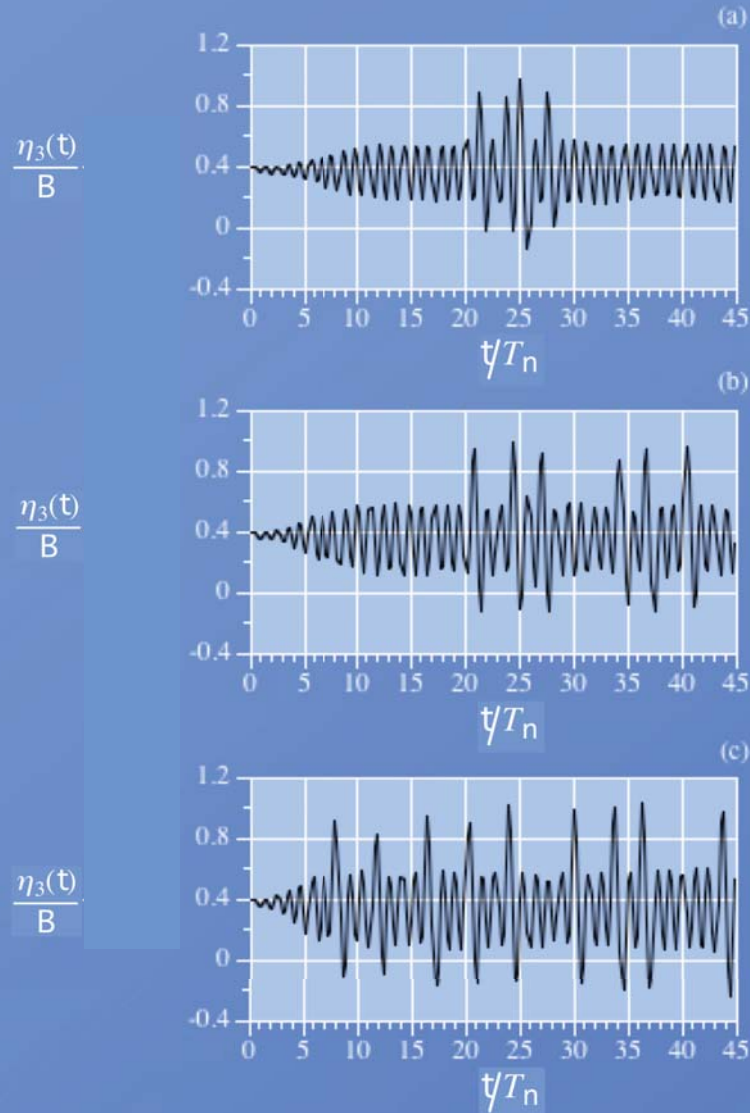
$$l_{cg}/B$$

(a) 2.09
(b) 2.03
(c) 1.98

$$C_v = 4.5$$

(Troesch and Hicks, 1994)





SIMULATION: MULTI-STATE PORPOISING

l_{cg}/B
 (a) 2.09
 (b) 2.03
 (c) 1.98

$C_V = 5.0$



EQUATIONS OF MOTION

Linear (unforced) (Martin, 1978)

$$[\mathbf{A}]\{\ddot{\eta}(t)\} + [\mathbf{B}]\{\dot{\eta}(t)\} + [\mathbf{C}]\{\eta(t)\} = \{0\}$$

Equivalent nonlinear model (unforced) (Hicks et al, 1994)

$$[\mathbf{A}]\{\ddot{\eta}(t)\} + [\mathbf{B}]\{\dot{\eta}(t)\} + [\overline{\mathbf{C}}(\eta(t))]\{\eta(t)\} = \{0\}$$

Equivalent nonlinear model (forced) (Hicks et al, 1994)

$$[\mathbf{A}]\{\ddot{\eta}(t)\} + [\mathbf{B}]\{\dot{\eta}(t)\} + [\overline{\mathbf{C}}(\eta(t))]\{\eta(t)\} = \{F(t)\}$$

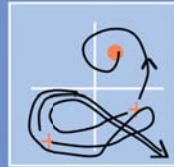
Fully nonlinear simulation (forced) (Zarnick, 1978,
and Akers, 1999, etc.)

$$[\mathbf{M}]\{\ddot{\eta}(t)\} = \{F(t)\}$$

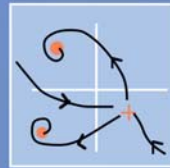


SCHEMATIC OF POSSIBLE HEAVE RAO

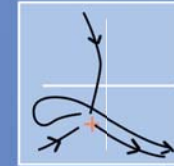
two saddles, one focus, tangles



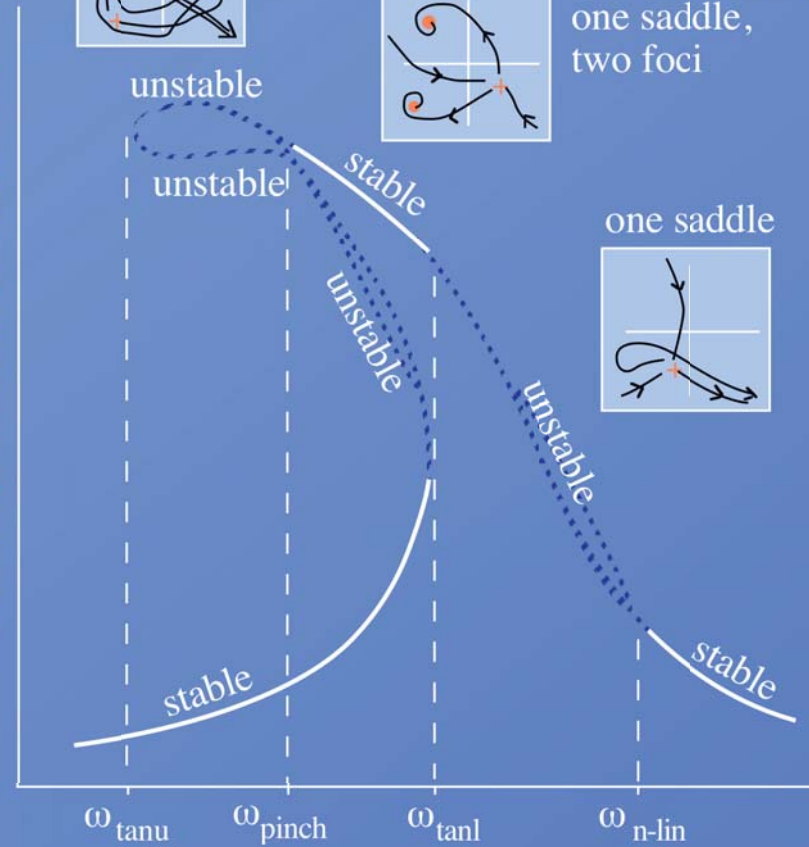
one saddle, two foci



one saddle

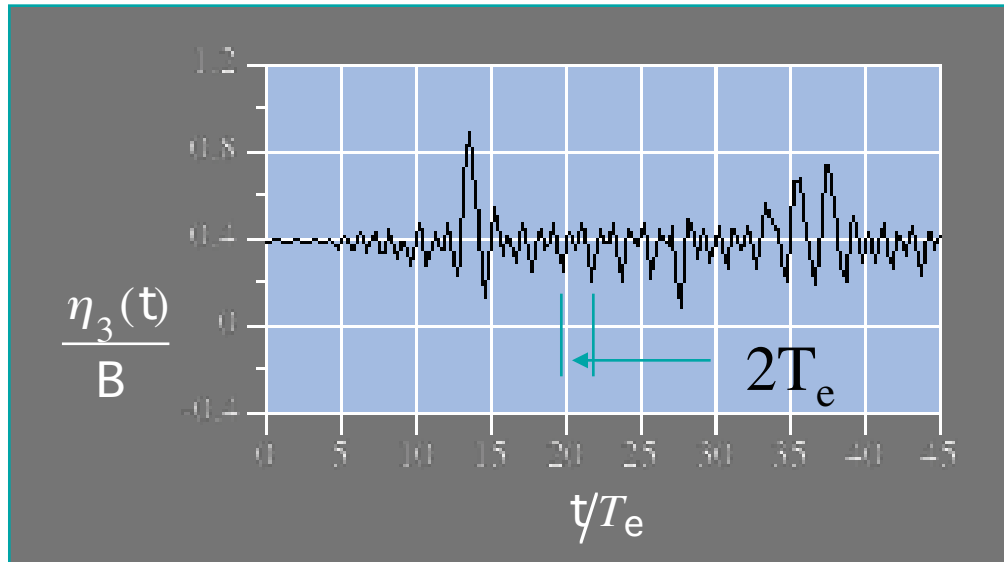


HEAVE RESPONSE FUNCTION



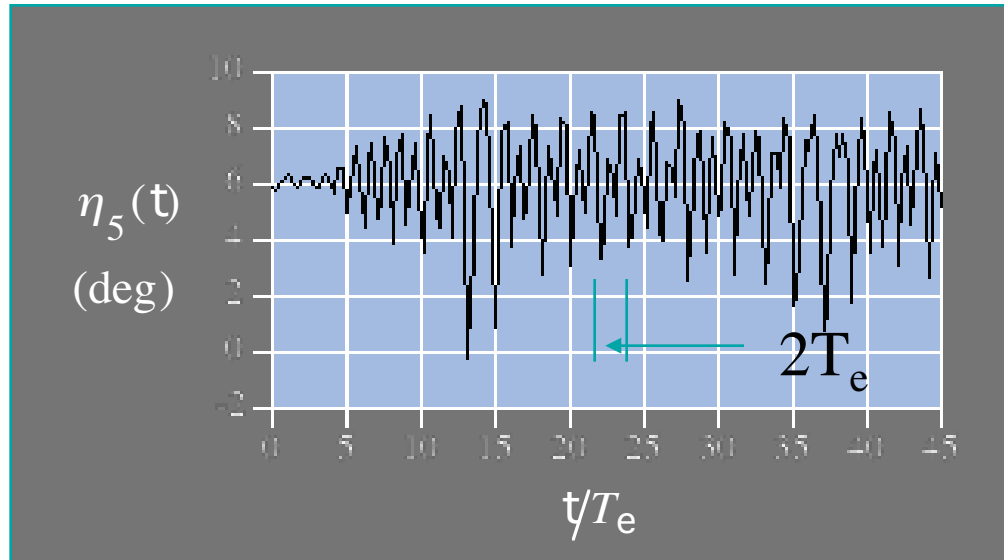
FREQUENCY





SIMULATED FORCED RESPONSE

(Troesch and Hicks, 1994)



$$C_v = 4.5$$

$$\zeta_0/B = 0.15$$

$$\lambda/L = 2.40$$

$$\omega_e \sqrt{B/g} = 2.29$$

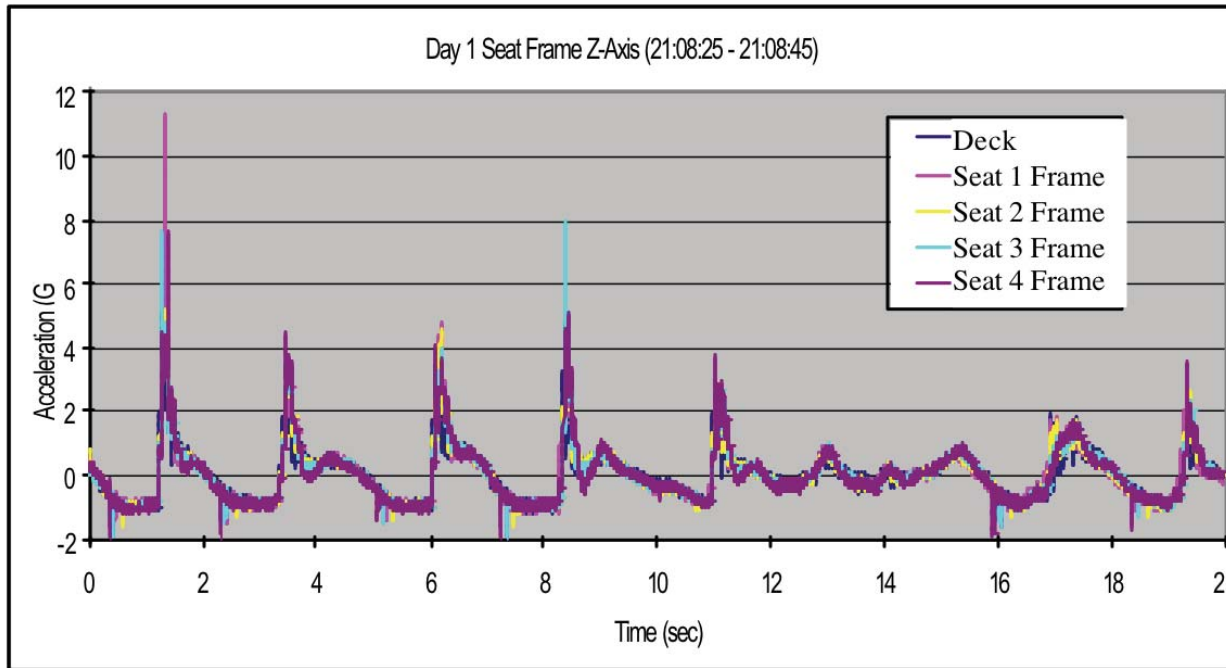


OUTLINE

- Planing Hull Hydrodynamics and Dynamics
Selected experiments and theories
with practical applications
- Nonlinear Dynamics Analysis Applied to Planing Hulls
- **Stochastic Vibro-Impact Model for Extreme Planing
Craft Acceleration Estimation**



Full Scale MARK V Acceleration Time Histories



MARK V deck & seat frame acceleration data collected January 10, 2003 by NSWC-PC.

Sea Trial conditions: 3.0 ft significant wave height, 8.3 second wave period



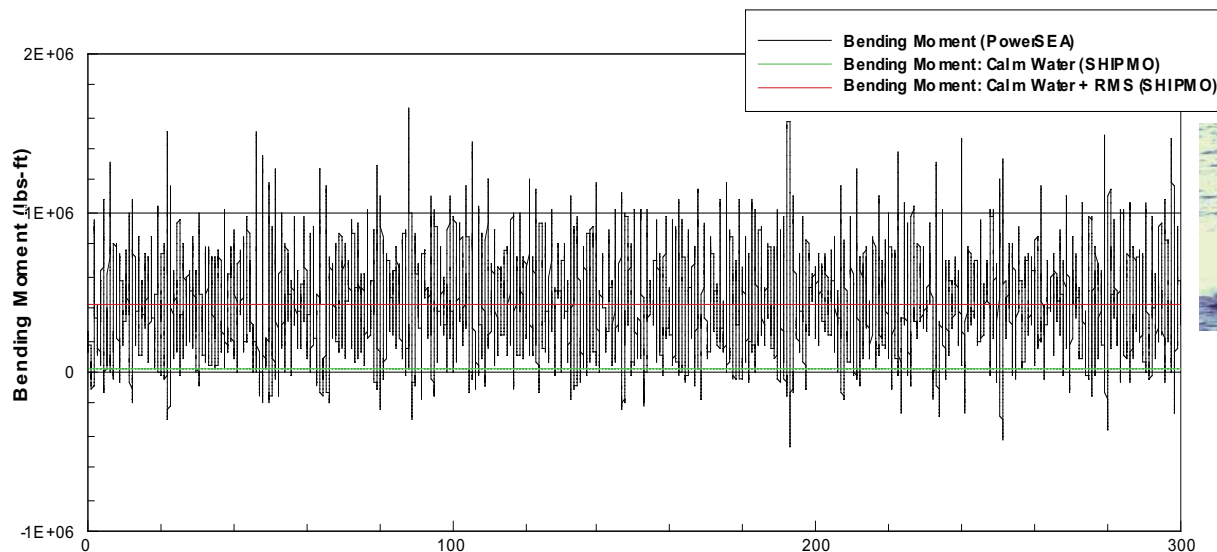
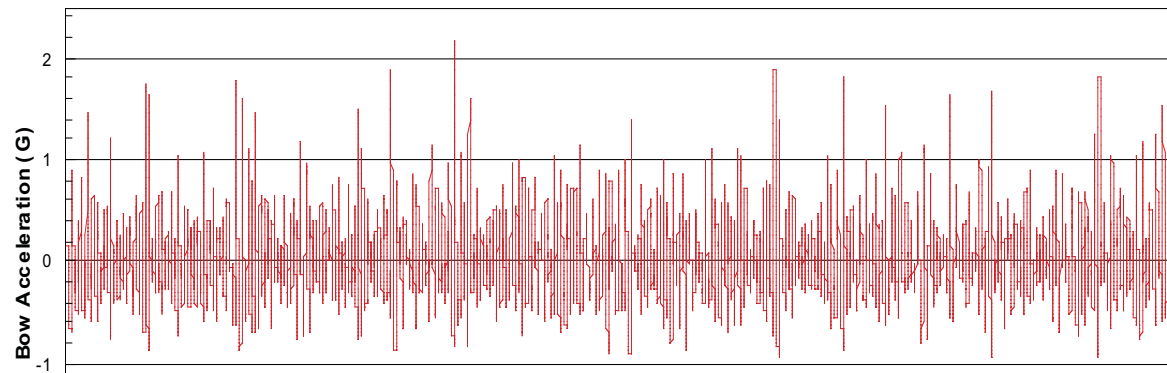
Armin Troesch
University of Michigan
Department of Naval Architecture and Marine Engineering

Dynamics and Hydrodynamics of High Speed Craft
June 29, 2010
Barranquilla, Colombia

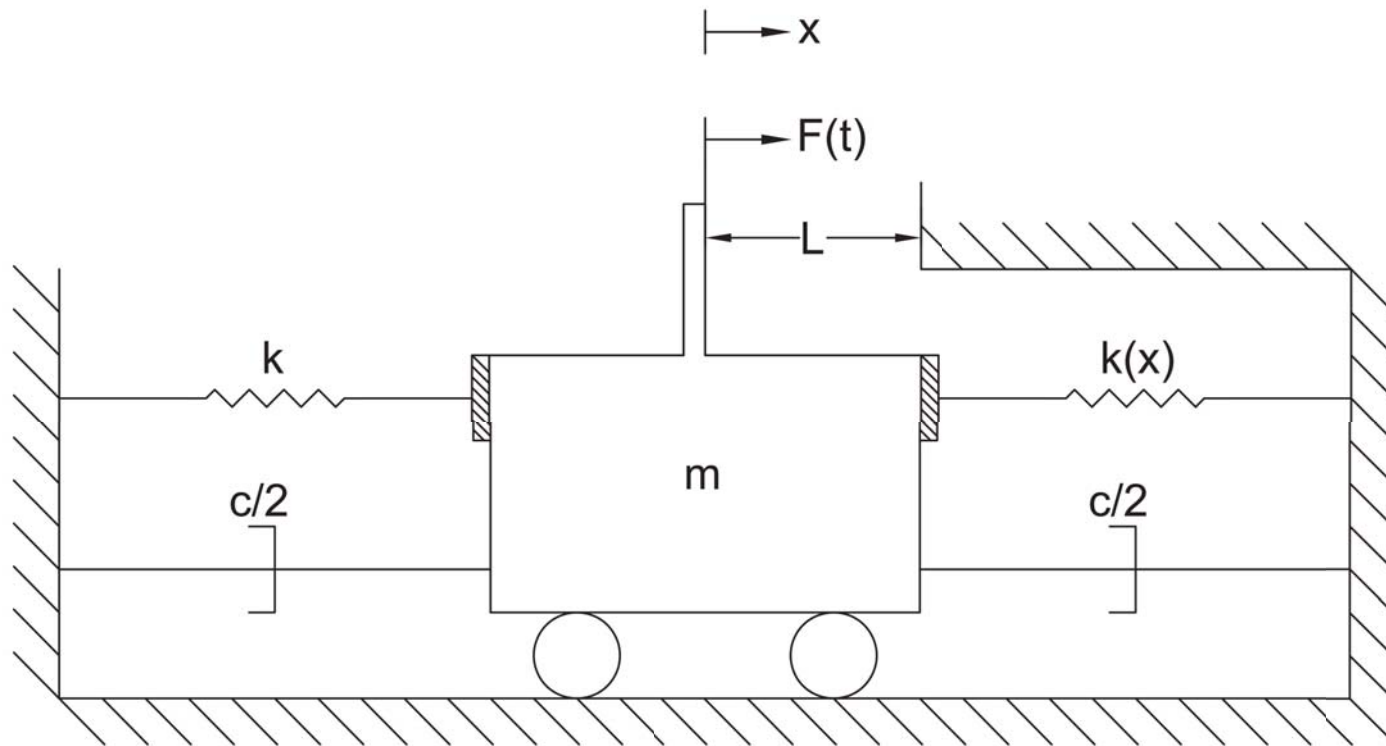
Extreme Responses in Stochastic Design Environments

Statistics of Impact

Simulated Mark V Acceleration at Bow and Bending Moment
(Sea State 2)

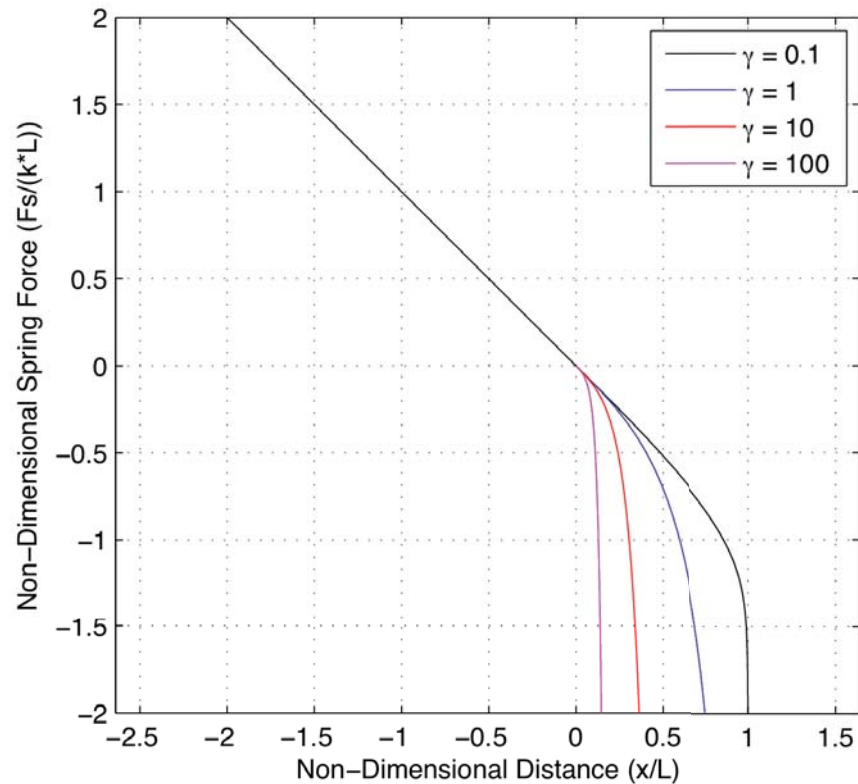


Stochastic Vibro-Impact Model for Extreme Planing Craft Acceleration Estimation

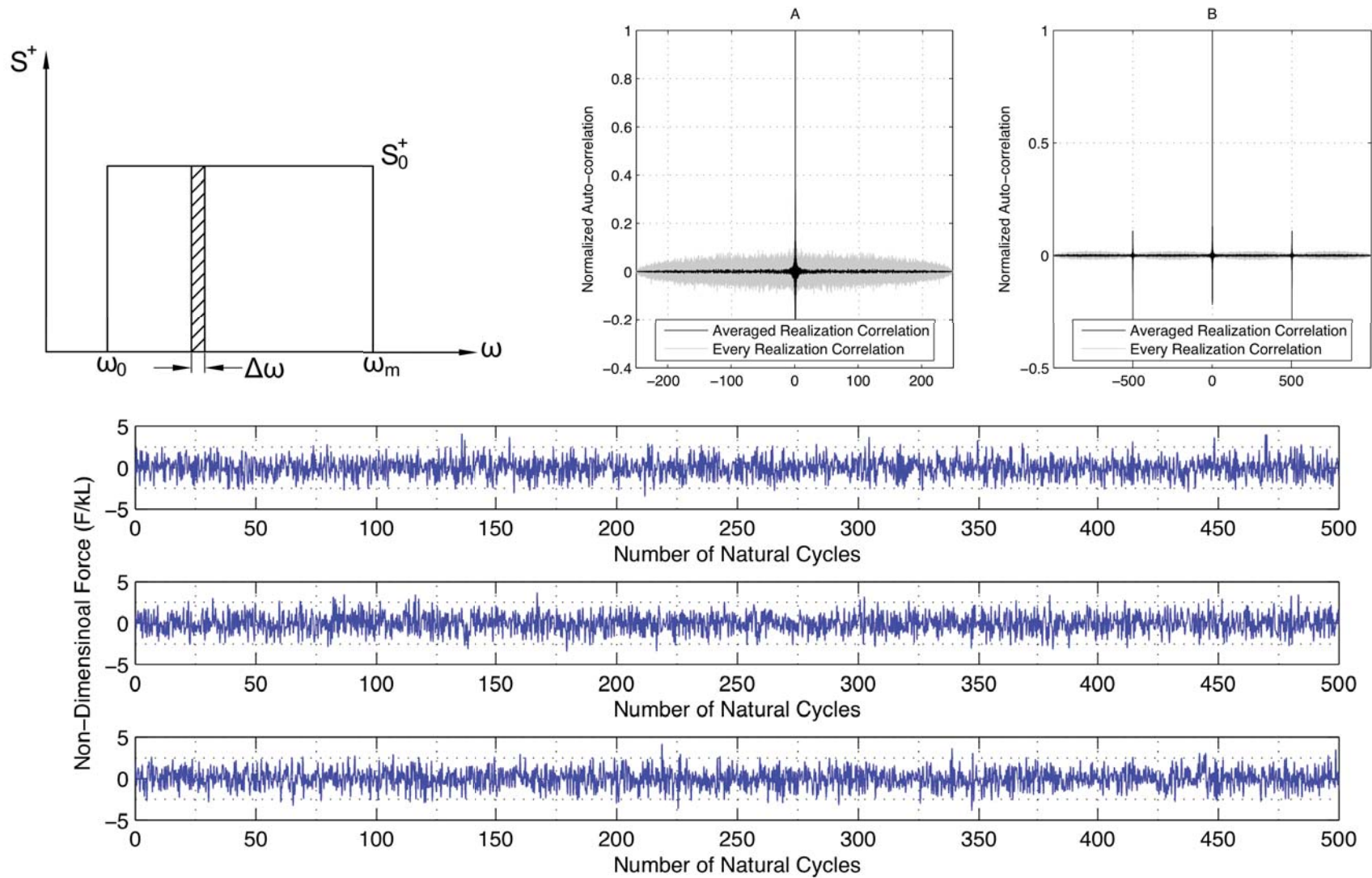


Stochastic Vibro-Impact Model Parametric Hardening Spring

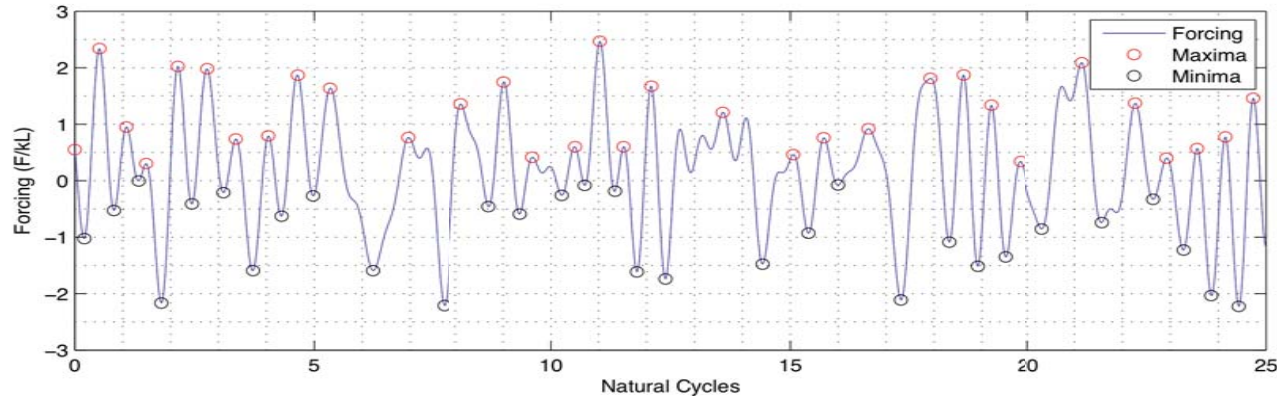
$$\begin{cases} Y > 0 & \frac{1}{4\pi^2}\ddot{Y} + \frac{\zeta}{\pi}\dot{Y} + Y \cos^{-\gamma}\left(\frac{Y\pi}{2}\right) \\ Y \leq 0 & \frac{1}{4\pi^2}\ddot{Y} + \frac{\zeta}{\pi}\dot{Y} + Y \end{cases} = \sum_{j=1}^N F_{\sigma} \sqrt{\frac{2}{N}} \cos(2\pi\eta_j\tau + \alpha_j)$$



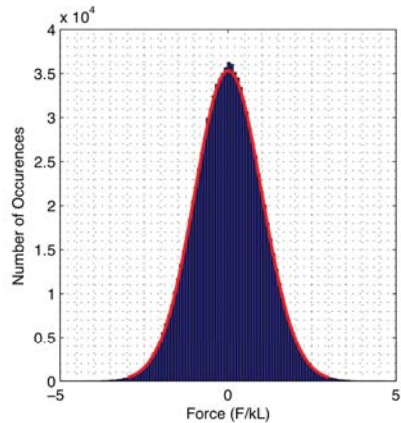
Ensemble of Band-limited Forcing



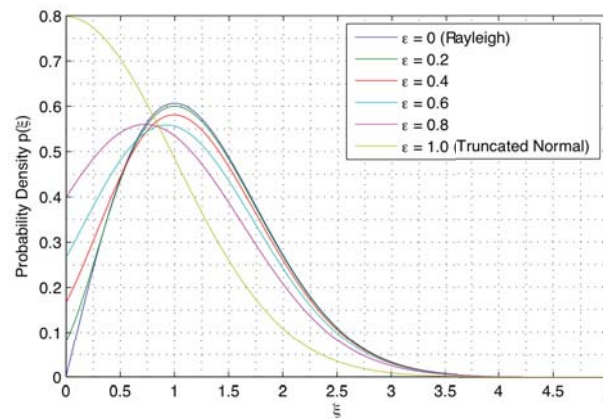
Extreme Value Statistics Background



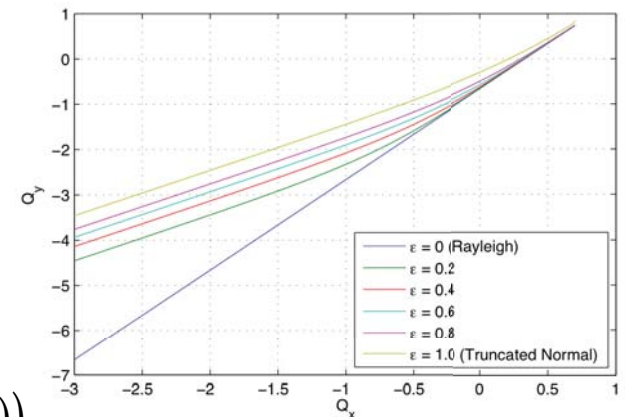
Elevation PDF



Maxima PDF



Quantile-Quantile Plot (Weibull Space)



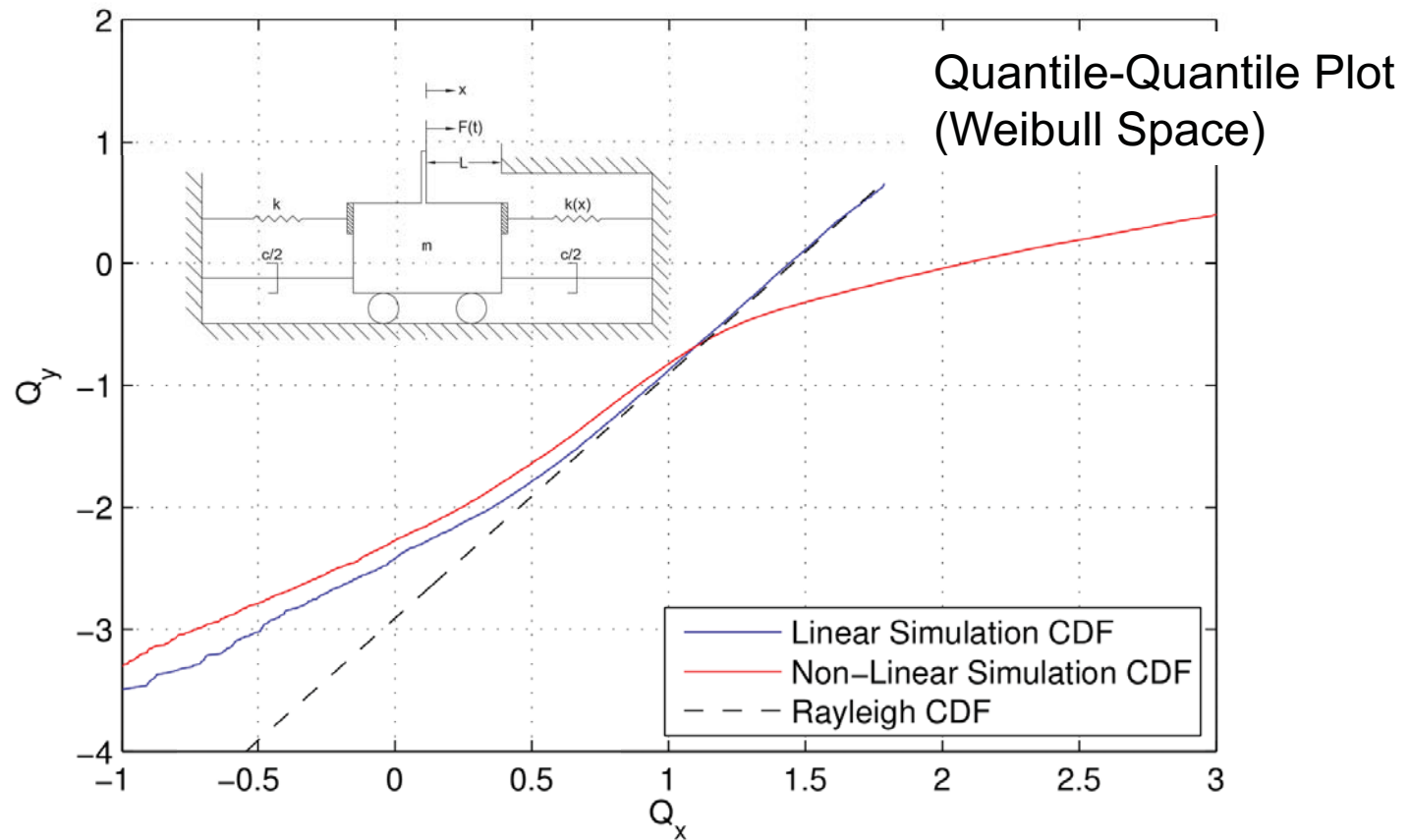
$$Q_x = \log_{10}(\xi)$$

$$Q_y = \log_{10}(-\log_{10}(1 - P(\xi)))$$



Stochastic Vibro-Impact Model

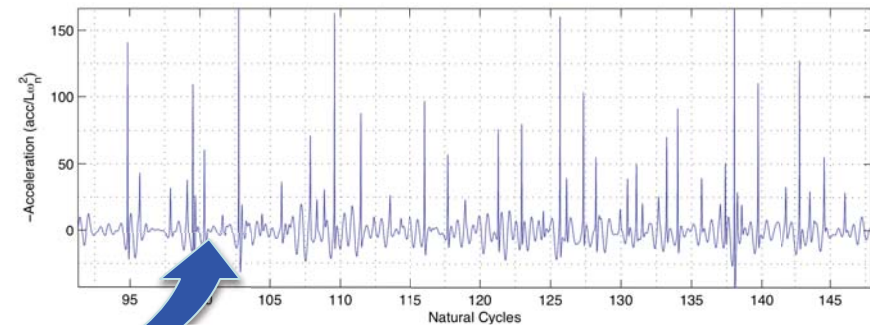
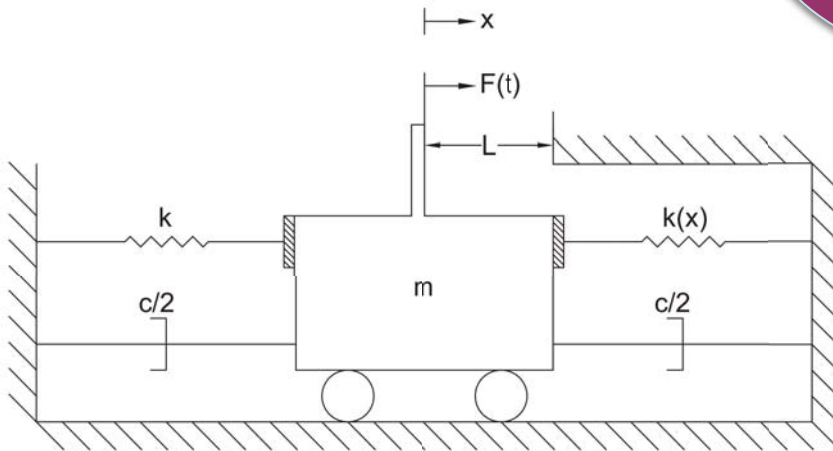
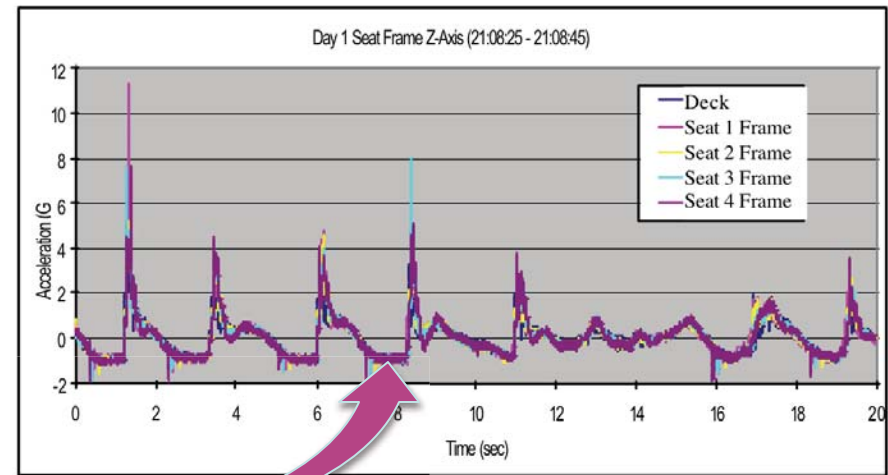
$$\xi=0.5, \underline{\eta}=0.5, \gamma=1.0, \varepsilon=0.5$$



$\xi=0.5$ (damping ratio), $\underline{\eta}=0.5$ (central forcing freq.), $\gamma=1.0$ (wall stiffness)



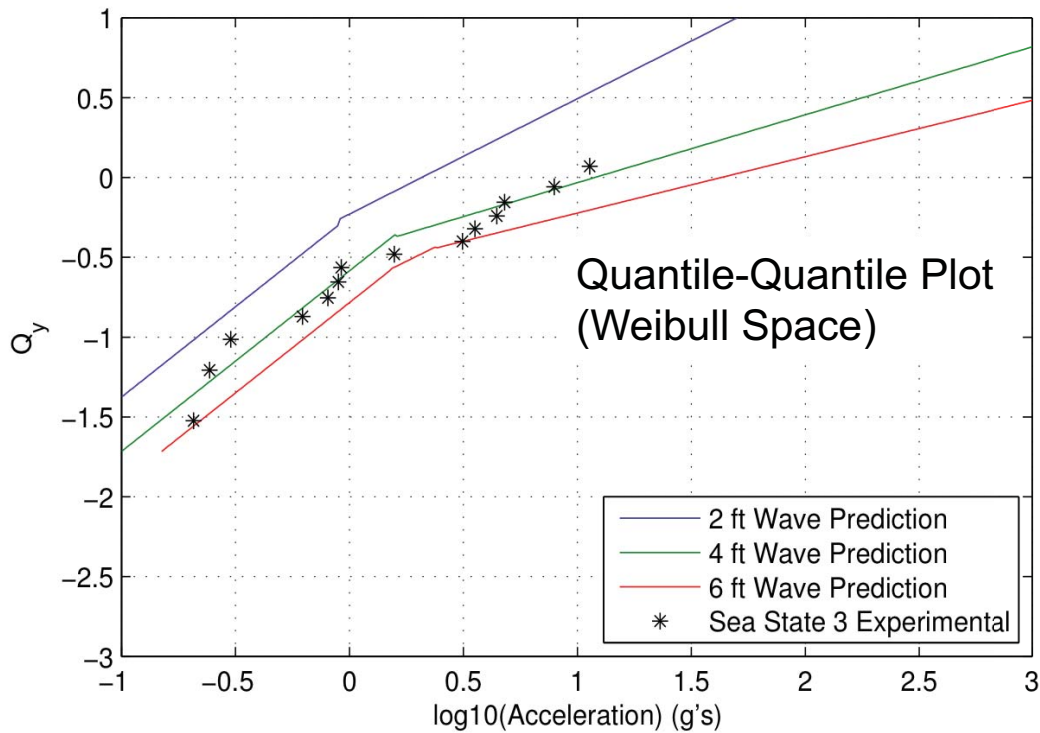
Stochastic Vibro-Impact Model for Extreme Planing Craft Acceleration Estimation - Application



Armin Troesch
University of Michigan
Department of Naval Architecture and Marine Engineering

Dynamics and Hydrodynamics of High Speed Craft
June 29, 2010
Barranquilla, Colombia

Stochastic Vibro-Impact Model for Extreme Planing Craft Acceleration Estimation



Extreme Acceleration Estimates

| $H_{1/3}$ | 11.5 g | 50 g | 20" | 2'30" |
|-----------|--------|--------|--------|-------|
| 2 ft | 95' | 600 yr | 2.1 g | 4.9 g |
| 4 ft | 19.3" | 2'18" | 11.8 g | 52 g |
| 6 ft | 8.5" | 23" | 42 g | 254 g |



Summary

- *Planing Hull Hydrodynamics and Dynamics*
Selected experiments and theories with practical applications
 - Complex coupled hydrodynamics and dynamics
 - Appendages play large role in assessing performance
 - Steady hydrodynamics relatively mature. Big challenge is resolving jet kinematics and geometry
 - Linear theory can be of use in assessing stability boundaries



Summary

- *Nonlinear Dynamics Analysis Applied to Planing Hulls*
 - Problem is rich in interesting dynamics and hydrodynamics
 - Reduced order nonlinear modeling required to understand global dynamics
 - Need to understand underlying physics to support brute force Monte Carlo simulations
 - Role that high fidelity simulators play in design and design optimization is not clear



Summary

- *Stochastic Vibro-Impact Model for Extreme Planing Craft Acceleration Estimation*
 - Traditional stochastic analysis assumptions for planing dynamics may be invalid
 - Extreme values of planing dynamics are the result of highly nonlinear, non-Gaussian processes
 - Reduced order models have important place in understanding extreme values





QUESTIONS ???

